2525. Control of elastic-hysteretic characteristics of multilayer all-metal vibration insulators with structural damping

Alexander S. Gvozdev¹, Vladimir S. Melentjev²

^{1, 2}Samara National Research University, Samara, Russian Federation
 ²Moskovskoe Shosse 34, Build 14, P.O. Box 443086, Samara, Russia
 ¹Corresponding author
 E-mail: ¹kipdla223@ssau.ru, ²vladamgenja@gmail.com
 Received 23 March 2016; received in revised form 25 January 2017; accepted 13 April 2017

DOI https://doi.org/10.21595/jve.2017.17016

Abstract. The use of all-metal vibration insulators is one of the most effective ways to eliminate the harmful effects of vibrations in mechanical systems. However, a similar type of parametrically controlled all-metal vibration insulators is little studied. In this research, the possibility of control of elastic-hysteretic characteristics of multilayer vibration insulators from metal is proved on the basis of the theory of similarity and theory of structural damping. It was found that the use of the square law of distribution of the friction coefficient on the height of the multilayer package and uniform compression of layers provides the high damping at resonance. The obtained results can be useful for design of effective parametrically controlled systems of vibration protection for any field of technology.

Check for updates

Keywords: elastic-hysteretic characteristics, all-metal controlled vibration insulator, structural damping, deformation amplitude, resonance, multilayer beam.

1. Introduction

There are many scientific works where a problem of energy dissipation in the systems with structural damping is studied. One of the most significant works is article of L. Goodman and J. Klamp [1-3]. In this paper, the authors suggested a way of hysteresis calculation, which occurs for bending two-layer console beam. The layers are pressed against each other by constant pressure. The beam rigidly fixed at one end. On the opposite end of the beam the force is applied. For this construction, the mathematic bilinear model of elastic-hysteresis loop was theoretically developed. As the authors note [1, 2, 4-7], the hysteresis is a result of an instant change of the inertia moment of the cross section of the beam at the certain points where the beam is deformed. The changes of the inertia moment of the cross section of the beam. The idea of forming the hysteresis, used into the mathematical model [1, 2, 8], is controversial. However, the authors have received accurate data: 1) the dependence of dissipated energy on the amplitude; 2) the presence of extremum on a graph of the dependence of dissipated energy from the contact pressure between the layers.

The similar results were obtained in the research of Kalinin for the multilayer beam compressed by the constant pressure [5, 9, 10]. The authors noted that the slippage of the layers occurs immediately in two-layer beam when a certain force on the end of beam is reached. In the multilayer beam the slip in each of the contact surfaces takes place simultaneously and according to the certain law the slip gradually spreads from surface to surface. It inhibits the growth of the energy dissipation coefficient ψ in the multilayer beam. The higher the energy dissipation coefficient ψ , the smaller is the amplitude in the resonance.

In [11-16] an extensive analysis of the opportunities of the different energy dissipation systems with the structural damping (e. g. an all-metal wire rope insulators, shock absorbers made of pressed wire material MR, multilayer dampers with various distribution of pressure between the layers) was carried out on the base of the theory of similarity and dimensionality. It was found that the multilayer structures with an instant slippage and the uniform pressure between the layers have the best damping properties. This finding has led to the development of some effective

designs of the damping devices for aviation and engineering industry.

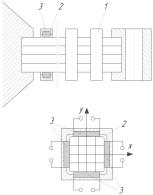
The principle of uniformity of compressive load along the length of the multilayer elastic element is the most simply implemented on the base of wire rope all-metal insulators [5, 17-19]. The elastic element in these vibration insulators consists of one or several wire ropes, tightly wrapped by wire. This elastic element connects two holders of the vibration insulator equipped with fixing bolts.

In [20] the design of a ribbon (instead of wire rope) vibration insulator is described. It is using of the similar idea for aerospace engineering. The uniformity of compression of layers in the package is provided by specially profiled elastic rings, which serve as the supports of the vibration insulator too. Experimental and theoretical study of these designs [20, 23] shown their high efficiency.

Unfortunately, in these designs [20, 21] the maximum value of the energy dissipation coefficient ψ_{max} is realized at the specific amplitude. Therefore, the value of the energy dissipation coefficient is not amenable to control during the process of vibrations. In this regard, the authors of this article had an idea to create the design of all-metal vibration insulators with controlled elastic-hysteretic characteristics. The pressure on the contact surfaces was selected as the parameter for the control of the properties of the all-metal multilayer vibration insulators with structural damping.

Several variants of the all-metal vibration insulators were developed on this principle [21, 24-27]. In one of them [27] (Fig. 1) the package of elastic elements of the square cross-sectional 1 is covered by clamps 2. Within the clamps 2 there are gaskets 3 made of electrostriction material that may change their transverse dimensions under the influence of electric voltage. The voltage is applied to the electrodes of the gaskets 3 from an external source of energy. Gaskets 3 allow changing of the pressure between the layers of the vibration insulator thus changing of the form of the hysteretic loop. Each of the two pairs of the opposite electrostriction elements are associated with independent voltage sources for separate control of the oscillation level in different directions. Voltage sources are controlled by the signals from the sensors of vibrations.

One of the variants of the design of the all-metal vibration insulator with perception spatial load is illustrated on Fig. 2. In the similar way, as in the first design, electrostriction items controlled by electrical voltage can be located under the clamps. Herewith on the mutually perpendicular areas of the vibration insulator the pressure on the layers can be individually controlled, depending on the vibration level in the corresponding direction.



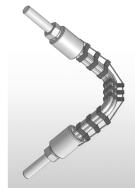


Fig. 1. The design of the all-metal vibration insulator with controlled elastic-hysteretic characteristics [27]

Fig. 2. The design of the controllable vibration insulator with perception spatial load [3, 27]

2. Materials and methods

It is known that the damping properties of the multilayer vibration insulators have a high level [27-30]. However, it can be improved.

2525. CONTROL OF ELASTIC-HYSTERETIC CHARACTERISTICS OF MULTILAYER ALL-METAL VIBRATION INSULATORS WITH STRUCTURAL DAMPING. ALEXANDER S. GVOZDEV, VLADIMIR S. MELENTJEV

An elastic-damping element is working on bending mainly. Elastic-damping element has a rectangular cross-section (Fig. 3). Zhuravskij D. I. derived an equation which describes the distribution of the shear stresses τ for a rectangular cross-sectional at bending [3, 31]:

$$\tau = \frac{3P}{2bH} \left[1 - \frac{4y^2}{H^2} \right],\tag{1}$$

where P is the transverse force in the considered section, N; b is width of a rectangular cross-sectional, m; H is height of a cross-sectional, m; y is the distance from the neutral axis to the corresponding contact surface, m. For multilayer beam H = hn, where n is the number of the layers in the multilayer package, h is the height of each layer.

The authors suggested that the dissipation energy coefficient will be maximized if the slippage will occur immediately on all contact surfaces. To achieve this, the height distribution of the friction coefficient f should be the same as the distribution of shear stresses [5, 9]:

$$\tau(y) = f(y)q,\tag{2}$$

where q is the compression pressure of the layers, N/mm².

By substituting Eq. (2) into Eq. (1) we find that on the contact surfaces of a multilayer package (Fig. 1, 2) need to create a pair of friction with the friction coefficient f, which obeys the following relationship:

$$f = \frac{3P}{2bhnq} \left[1 - \frac{4y^2}{n^2 h^2} \right],\tag{3}$$

where P is the amplitude of the transverse force at which delaminating of the multilayer package is provided, N. The friction coefficient f reduces roundtrip from the neutral axis (Fig. 3).

When the level of tangential stresses equal to the level of the distributed friction forces $(q_0 = f_0)$, the simultaneous slippage of all the layers in the multilayer package will happen [25].

At the bending of such an elastic-hysteresis element (q = const) the hysteretic loop will be in the form of parallelogram (Fig. 4) with two characteristic stiffness one of which is the initial stiffness (delaminated stiffness), defined by the expression:

$$C_0 = \mathrm{tg}\beta = \frac{\mu E n^3 J}{l^3}.$$
(4)

And the other is laminated stiffness, defined by the expression:

$$C_p = \mathrm{tg}\alpha = \frac{\mu E n J}{l^3}.$$
(5)

where α is the angle between the horizontal axis x and the site of the load curve which corresponds to the laminated multilayer beam; β is the angle between the horizontal axis x and the site of the load curve which corresponds to the delaminated multilayer beam; μ is the coefficient that depend on the boundary conditions of the displacement of the movable end of the multilayer beam; *E* is the modulus of elasticity of material, N/m²; $J = (bh^3)/12$ is the inertia moment of the transverse section of the layer in the multilayer package, m⁴; *l* is the length of the beam, m.

2525. CONTROL OF ELASTIC-HYSTERETIC CHARACTERISTICS OF MULTILAYER ALL-METAL VIBRATION INSULATORS WITH STRUCTURAL DAMPING. Alexander S. Gvozdev, Vladimir S. Melentjev

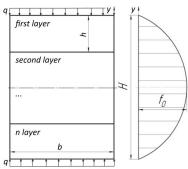


Fig. 3. The cross-section and the graph of distribution of the values of the friction coefficient on the thickness of the multilayer beam

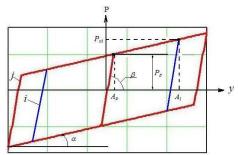


Fig. 4. The elastic-hysteretic characteristic of the multilayer all-metal controlled vibration insulator 1

The energy dissipation coefficient ψ is determined as:

$$\psi = \frac{16bqhf_0(1-1/n)}{3P},$$
(6)

where f_0 is the maximum friction coefficient of the friction pair on the neutral axis of the multilayer package (Fig. 3). This equation shows that the value of energy dissipation coefficient ψ may be controlled by changing of the value of compression loads q. To ensure the permanence of energy dissipation coefficient $\psi = \psi_0 = const$, which does not depend on the force P amplitude, the authors derived the following relationship for the compression loads:

$$q = \frac{3P\psi_0}{16bhf_0 \left(1 - \frac{1}{n}\right)}.$$
(7)

Since the amplitude of the force P acting on the vibration insulator depends on the acceleration of the object of the vibration protection, it is easy to implement dependence Eq. (7) using a simple electronic circuit with an acceleration sensor placed on the object of the vibration protection.

Relationship Eq. (3) of distribution of the friction coefficient on the height of the multilayer package f(x, y) with the maximum f_0 on the neutral axis can be realized with the help of various coverings. Thus, on the basis of the all-metal vibration insulators with structural damping the adjustable systems of the vibration protection can be created.

Let us consider the possibilities of the created model of the adjustable vibration insulator. We define the initial parameters of the system. We have a multilayer steel beam consisting of n layers, height H = const. The height of each layer is h, the width is b, and the length is l. We accept the law Eq. (3) of distribution of the friction coefficient on the thickness of the multilayer beam with the maximum of f_0 on the neutral axis (Fig. 3). Let suppose that the surface of each layer is treated

so that the friction coefficient f on the neutral axis of the multilayer beam is $f_0 = 0.15$. The pressing of the layers of the multilayer beams to each other is provided by the distributed load q. The multilayer beam is loaded on its end by the concentrated force P.

Taking value $\mu = 3$ for selected boundary conditions, we denoted the stiffness parameters of hysteresis as:

$$C_p = \operatorname{tg}\alpha = \frac{3E \cdot J_p}{l^3},\tag{8}$$

$$C_0 = tg\beta = \frac{3L^2 f_0}{l^3},$$
(9)

where J_P , J_0 are the inertia moments of the cross-section of the laminated and delaminated multilayer beam (Fig. 4).

The laminated moment of inertia is (the layers are deformed independently):

$$J_p = \sum_{i=1}^n \frac{b \cdot h_i^3}{12} = n \frac{b \cdot h^3}{12}.$$
(10)

The delaminated moment of inertia is (the layers are deformed together):

$$J_0 = \frac{b \cdot H^3}{12} = n^3 \frac{b \cdot h^3}{12}.$$
 (11)

Let us determine the force Eq. (12) and the deflection Eq. (13), at which the layers of the beam begin to slip:

$$P_p = \frac{2}{3}bHf_0q,\tag{12}$$

$$A_p = \frac{P_p l^3}{3EJ_0}.$$
 (13)

Then the maximum force of the separate hysteretic loop with amplitude A is given by:

$$P_{0} = \begin{cases} A \cdot C_{H} \cdot sign(y), & A \leq A_{p} \\ \left[(A - A_{p}) \cdot C_{p} + P_{p} \right] \cdot sign(y), & A > A_{p}. \end{cases}$$
(14)

We find the parameters derived from the hysteretic loop. It is the mean-cyclic stiffness:

$$C = \frac{P_0}{A}.$$
 (15)

And the dissipated cyclic energy W, which is equal to the area of the hysteretic loop with arbitrary amplitude A:

$$W = 4A \cdot P_0 - 4P_p \cdot A_p - 4(P_0 - P_p) \cdot (A + A_p).$$
⁽¹⁶⁾

The energy dissipation coefficient ψ is determined as:

$$\psi = \frac{2W}{AP_0}.$$
(17)

By input of the specific value of the parameters and using Eqs. (3) and (17), we obtained the dependence of the energy dissipation coefficient ψ on the deformation amplitude A and the number of layers of the multilayer beam (Fig. 5):

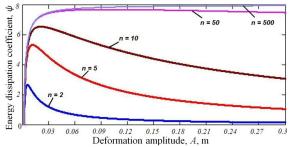


Fig. 5. The dependence of the energy dissipation coefficient ψ from the deformation amplitude A. The number of layers in the multilayer beam is fixed. In this, l = 1 m, b = 0.1 m, H = const = 0.1 m, q = 10 MPa, $f_0 = 0.15$

Substituting Eqs. (10) and (11) to Eq. (14), with $n \to \infty$ it is easy to find the limit of the function Eq. (17), which turns out to be $\psi_{max} = 8$ at the optimum deformation amplitude:

$$A_{optim} = A_p \cdot \left(1 + \sqrt{\frac{P_p}{A_p \cdot C_p}} \right).$$
⁽¹⁸⁾

Dependence Eq. (18) deformation amplitude A_{optim} on the number of layers n of the multilayer beam is the linearly increasing function.

We investigated the dependence of the energy dissipation coefficient ψ on the number of layers *n* of the multilayer beam with the optimum deformation amplitude A_{optim} . We represent the maximum value of the force corresponding to the optimum deformation amplitude A_{optim} as:

$$P_{max}^{optim} = \frac{(A_{optim} - A_p) \cdot S}{k^2} + P_p.$$
⁽¹⁹⁾

And the area of the hysteresis loop for the optimum deformation amplitude A_{optim} as:

$$W_{max}^{optim} = 4A_{optim} \cdot P_{max}^{optim} - 4P_p \cdot A_p - 4(P_{max}^{optim} - P_p) \cdot (A_{optim} + A_p).$$
(20)

Then the energy dissipation coefficient at the optimum deformation amplitude is represented by the expression:

$$\psi_{max}^{optim} = \frac{2W_{max}^{optim}}{A_{optim} \cdot P_{max}^{optim}}.$$
(21)

Next, we defined the dependence of the scattered energy on the compressive pressure q at a deformation amplitude $A_{max} = const$ and n = const. To do this, we substitute in the Eq. (14), the expression (12) of the force of the lamination P_p (Fig. 4). We previously denoted a set of constants as Z:

$$Z = \frac{2}{3}bHf_0.$$
 (22)

The force P_0 at a given value of the compressive pressure q is:

$$P_{0} = \begin{cases} A \cdot C_{H}, & A \leq A_{p}, \\ \left(A - \frac{Z \cdot q}{C_{H}}\right) \cdot C_{p} + Z \cdot q, & A > A_{p}. \end{cases}$$
(23)

We substitute in the Eq. (16), the Eq. (23). The dependence of the area of the hysteretic loop on the magnitude of the compressive pressure q is:

$$W_{q} = 4A \cdot P_{q} - 4 \frac{Z^{2} \cdot q^{2}}{C_{H}} - 4(P_{q} - Z \cdot q) \cdot \left(A + \frac{Z \cdot q}{C_{H}}\right).$$
(24)

Next, we defined the maximum compressive pressure q_{max} . To obtain the result we equated the left-hand side of the Eq. (24) to zero and solved the resulting quadratic equation for q. At the same time, we got rid of the function *if* in the Eq. (23) of the force *P*. We accept the condition that if W = 0, then the force *P* is calculated by the delaminated stiffness C_0 Eq. (9) because sliding does not occur. Then the maximum value of the compressive pressure q_{max} at which the sliding at a given deformation amplitude *A* does not occur is defined as:

$$q_{max} = \frac{3}{8} \frac{A E H^2}{f_0 l^3}.$$
(25)

The optimum pressure q_{optim} is a half of the maximum load q_{max} [3, 31]:

$$q_{optim} = \frac{3}{16} \frac{A \cdot E \cdot H^2}{f_0 \cdot l^3}.$$
 (26)

We defined the compressive pressure q at the extremum point of function $\psi = f(q)$, where the energy dissipation coefficient ψ is maximal. To obtain the result, we differentiated the function $\psi = f(q)$ for q and equated to zero the derivative function. We obtained a quadratic equation, solving which we found:

$$q_{ekstrem}^{\psi} = \frac{A \cdot C_{\rm H}}{Z \cdot (3 - n^2)} \cdot \left[1 - \sqrt{1 - (3 - n^2) \cdot \left[\frac{1}{n^2} \cdot \left(\frac{Z}{C_{\rm H}} - 1\right) + 1\right]} \right].$$
(27)

Next, we defined the dependence of the energy dissipation coefficient ψ on the deformation amplitude A and on the number of layers n of the multilayer beam as:

$$\psi(A,n) = \frac{P_0 \cdot A - P_p \cdot A_p - (P_0 - P_p) \cdot (A + A_p)}{0,125A \cdot P_0},$$
(28)

where:

$$P_{0}(A,n) = \begin{cases} A \cdot C_{H}, & A \le A_{p}, \\ \frac{(A-A_{p}) \cdot C_{H}}{n^{2}} + P_{p}, & A > A_{p}. \end{cases}$$
(29)

The study of the dynamic characteristics of the mechanical system is conveniently carried out on the basis of the amplitude-frequency characteristic (AFC). The plot of the AFC presented in the dimensionless form (Fig. 7). It is necessary to determine the dependence of the dynamicity coefficient η on the relative frequency of the mechanical system ν . It is known that the dynamicity coefficient η depends on the damping coefficient γ . For a given mechanical system the damping coefficient γ depends on the deformation amplitude A [5, 27, 32]. We calculated the average damping coefficient γ_{avg} as a function of the number of layers *n* of the multilayer beam to eliminate this dependence within the system. This coefficient does not depend on the deformation amplitude. Therefore, we will use them in further calculations of the dynamics of the mechanical system. The average damping coefficient is defined as [27, 32]:

$$\gamma_{avg} = \frac{\psi_{avg}}{2\pi}.$$
(30)

We obtained an average energy dissipation coefficient ψ_{avg} as:

$$\psi_{avg} = \frac{S_{\psi}}{A},\tag{31}$$

where S_{ψ} is the area under the graph of the dependence of the energy dissipation coefficient ψ on the deformation amplitude A.

The functional dependence of the dynamicity coefficient η on the dimensionless frequency ν is given by [5, 33]:

$$\eta(\nu, \gamma) = \sqrt{\frac{1 + \gamma^2}{(1 - \nu^2)^2 + \gamma^{2'}}}$$
(32)

where the dimensionless frequency ν is defined as:

$$\nu = \frac{\omega}{\omega_0},\tag{33}$$

where the natural circular frequency of the system is:

$$\omega_0 = \sqrt{\frac{C_p}{m}},\tag{34}$$

where m is the mass of the oscillating object.

The values of the energy dissipation coefficient ψ for multilayer beam with the variable friction coefficient f(y) and the values of ψ for multilayer beam with the equal f [34] on all the contact surfaces are compared. Eskin method allows the calculation of multilyesr beam only with an even number of layers n. The number of layers n should be greater than four. Comparison was made at the following parameters: b = 0.1 m, H = 0.1 m, L = 1 m, $f_0 = 0.15$, $q = 10^7 \text{ H/m}^2$, $E = 2.1 \cdot 10^{11} \text{ H/m}^2$. The comparison was made at similar thickness of the hysteretic loops W for each of the layers n. The comparison results are presented in Fig. 6.

3. Results

As a result of our research the following was found.

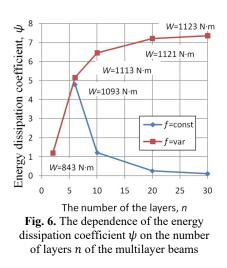
1. The damping coefficient of the multilayer all-metal vibration insulators (for example, Fig. 1, 2) can be increased. To obtain the result it is necessary to create a pair of friction with the quadratic distribution according to Eq. (3) of the friction coefficient on the contact surfaces of the multilayer beam (Fig. 3).

2. In the multilayer all-metal vibration insulators with the law of the quadratic distribution of the friction coefficients between layers exists optimal deformation amplitude Eq. (18), at which the energy dissipation coefficient ψ is maximal and can be represented by the Eq. (21).

3. The magnitude of compressive pressure q can control the level of the energy dissipation coefficient ψ . To ensure the stability of the energy dissipation coefficient $\psi = \psi_0 = const$ at change the force P (or the deformation amplitude A), it is necessary to ensure the law of distribution of the compressive pressure as Eq. (7).

4. The optimum pressure q at which the area of the hysteretic loop is maximal defined from the Eq. (26).

5. The value of the compressive load q, required to achieve the maximum value of the coefficient ψ , can be got from the Eq. (27).



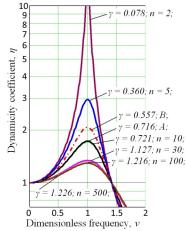


Fig. 7. The dependence of dynamicity coefficient η on the dimensionless frequency ν : *A* – corrugated damper (avg), *B* – wire pressed materials (avg)

6. The energy dissipation coefficient ψ for multilayer beam with f = variable is larger than for a beam with the uniform f = const when n = idem and W = idem (Fig. 6). For beam with f = const energy dissipation coefficient ψ decreases with increasing number of layers n due to lack of control. For beam with f = variable energy dissipation coefficient ψ increases with n and approaches to the maximum $\psi_{max} = 8$ for systems with structural damping.

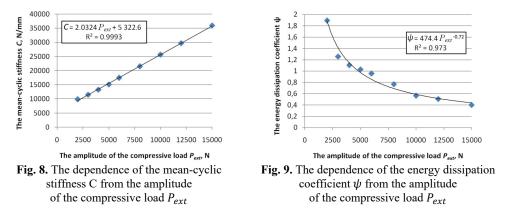
7. To evaluate the effectiveness of the vibration isolation systems based on the different elastic-damping elements (e.g. corrugation, pressed wire material MR) we can use dependence of the dynamicity coefficient η on the dimensionless frequency ν (Fig. 7).

Ability to control stiffness and damping of the multi-layer-metal damper via magnitude compressive load has been proven experimentally [27, 31]. Static tests of the damper were carried out in the mode uniaxial loading. The obtaining of a set of the hysteresis loops when exposed to cyclic compressive force with amplitude on the damper is the basis of experiment P_{ext} . The load P_{ext} at the same time compresses the layers and causes a shift between the layers of the damper (sliding between layers). The experimental procedure is described in detail in the source [5, 31]. For the experiment was taken a multilayer steel damper with the following parameters: the number of layers in the package n = 19, the height of the package H = 9.45 mm, the package width b = 33 mm. The experimentally obtained dependencies of the mean-cyclic stiffness C and the energy dissipation coefficient ψ from the load P_{ext} are shown in Fig. 8 and Fig. 9. R^2 is the value of reliability of approximation.

It is clearly seen that the mean-cyclic stiffness C increases as the load P_{ext} increases, and the

energy dissipation coefficient ψ decreases. This behavior of the parameters is consistent with the calculations given in this paper. In [35, 36] experimentally studied the different coating for plates of frictional dampers with provided constant compressive load (created due to the springs). Researches were carried out at different frequencies and amplitudes. In [37] investigated experimentally the multilayer damper under different loads and different sliding speeds. In [31, 38, 39] was experimentally investigated behavior of the friction coefficient of different pairs of metal strips (of copper, gold, and steel). The effect of the normal load and sliding speed on the coefficient of friction has been investigated.

These studies confirm the possibility of adjusting the hysteretic characteristics of the plate damper by choosing the friction coefficients between the layers and the control of the compressive load.



4. Conclusions

The analysis of the functional dependence Eq. (17) shown in Fig. 5 shows that the energy dissipation coefficient ψ is practically constant on the operating frequency of the vibration protection system at the number of layers of the elastic-damping element (multilayer all-metal beam) n > 50. Further increase of the number of layers of the multilayer beam gives a slight increase of the coefficient ψ . The analysis of the graphs presented in the article, shows that the proposed principles allows to create a fairly simple design that provides an almost constant damping coefficient in a wide range of frequencies and deformation amplitudes of the all-metal vibration insulator. Moreover, at the small deformation amplitudes (in the after resonance region) of the vibration isolation system, the damping coefficient is close to zero, that reduces losses.

It should be noted that we determined the compressive load where the energy dissipation coefficient ψ is maximal Eq. (21) by differentiation the function $\psi = f(q)$ for q and the equating the derivative to zero Eq. (27).

The analysis of the Eq. (26) shows that for n = const, the optimal compressive pressure q_{optim} depends only on the deformation amplitude. And it is a linearly increasing function. This fact allows using the dependence Eq. (26) to design of the vibration isolation systems with the simple parametric control of the elastic-hysteretic characteristics.

The graph of the dynamicity coefficient η on the dimensionless frequency ν (Fig. 7) indicates the advantage of using multilayer all-metal vibration insulators compared with the similar design on the basis of the wire pressed material "metal analogue rubber (MR)" or package of the corrugated tapes at designing parametrically controlled elastic-hysteretic systems. For example, for a beam from 100 layers (each layer is represented by a thin metal plate, the plates have equal thickness) the dynamicity coefficient η is 1.127. It corresponds to an increase of the amplitude on only 30 percents at the moment of the resonance. While the pressed wire material MR provides a twofold increase of the amplitude. Thus, in the course of this research it was found that the multilayer all-metal vibration insulator with simultaneous sliding of the all plates in the package (in which a quadratic distribution of the friction coefficient f along the height of the cross-section and the uniform distribution of the compression q is implemented) provides the energy dissipation coefficient ψ higher than the vibration insulator with a constant friction coefficient between the layers provides. Furthermore, the increasing of the number of layers n leads to an increase of the stability of the energy dissipation coefficient ψ as a function of the deformation amplitude A. As a result, even if the package has only a few tens layers, the energy dissipation coefficient is approaching the theoretical limit $\psi_{max} = 8$. Moreover, within the range of operating frequencies the coefficient ψ increases proportionally to the deformation amplitude A (Fig. 5), providing the minimum or zero value of the coefficient ψ for the small amplitudes A. It increases the efficiency of the vibration isolation in the range of after resonance frequency (zone of protection from vibration).

Thus, the design, based on the principles proposed in the article, has a linear control law, provides low amplitude at resonance, and effective work in the range of after resonance frequency.

Acknowledgements

This work was supported by the Ministry of Education and Science of the Russian Federation in the framework of the implementation of the Program "Research and Development on Priority Directions of Scientific-Technological Complex of Russia for 2014-2020".

References

- Ao H., Jiang H., Ulanov A. Estimation of the fatigue lifetime of metal rubber isolator with dry friction damping. Experimental Mechanics in Nano and Biotechnology, Parts 1 and 2, Vol. 326, Issue 328, 2006, p. 949-952.
- [2] Ao H., Jiang H., Wei W., et al. Study on the damping characteristics of MR damper in flexible supporting of turbo-pump rotor for engine. 1st International Symposium on Systems and Control in Aerospace and Astronautics, Vols. 1-2, 2006, p. 618-622.
- [3] Ponomarev Yu, Kalakutsky V. The Multilayer All-Metal Vibration Insulators with Elastic Elements of a Regular Structure (Mnogosloinie Celnometallicheskie Vibroizoliatori s Uprugimi Elementami Reguliarnoi Strukturi). Samara, Samara State Aerospace University, 2003, p. 198.
- [4] Melentjev V., Gvozdev A. Method of static characteristics calculation of adjustable two-ring type full-metal vibroinsulators in a non-linear approach. Modern Applied Science, Vol. 9, Issue 2, 2015, p. 267-278.
- [5] Popov E. The Theory and Calculation of the Flexible Elastic Rods (Teoria i Raschet Gibkih Stergnei). Moscow, Nauka, 1986, p. 296.
- [6] Nurkan Y., Emir S. Vibrations of a rectangular bridge as an isotropic plate under a traveling full vehicle model. Journal of Vibration and Control, 2006, https://doi.org/10.1177/1077546306061411.
- [7] **Popp K., Panning L., Sextro W.** Vibration damping by friction forces: theory and applications. Journal of Vibration and Control, Vol. 9, 2003, p. 419-448.
- [8] Mihir C. Free vibration of tapered isotropic rectangular plates. Journal of Vibration and Control, 2012, https://doi.org/10.1177/1077546310396800.
- [9] Balyakin V., Falaleev S., And Novikov D. Hermeticity of secondary gas end seal assembly. Gazovaya Promyshlennost, Vol. 8, 2002, p. 56-58.
- [10] Srinivasan A., MacFarland D. Characterization of laws of friction in the context of engine blade dynamics. Journal of Engineering for Gas Turbines and Power, Vol. 120, Issue 4, 1998, p. 759-765.
- [11] Barry O., Zu J., Oguamanam D. Analytical and experimental investigation of overhead transmission line vibration. Journal of Vibration and Control, 2014.
- [12] Hongrui A., Degang H., Hongyuan J. Modeling of micro-friction in a multi-layer metal plate damper for a rotor support system. 2nd International Symposium on Systems and Control in Aerospace and Astronautics, Shenzhen, China, 2008, https://doi.org/10.1109/ISSCAA.2008.4776322.
- [13] Jiang H.-Y., Zhang R.-H., Zhao K.-D., et al. Analysis of damping characteristics of squeeze film damper with metal rubber. Tuijin Jishu/Journal of Propulsion Technology, Vol. 26, Issue 2, 2005, p. 174-177.

- [14] Jiang H.-Y., Hao D.-G., Xia Y.-H., et al. Damping characteristics calculation method of metal dry friction isolators. Journal of Beijing Institute of Technology (English Edition), Vol. 17, Issue 2, 2008, p. 173-177.
- [15] Krack M., Tatzko S., et al. Reliability optimization of friction-damped systems using nonlinear modes. Journal of Sound and Vibration, Vol. 333, Issue 13, 2014, p. 2699-2712.
- [16] Lazutkin G., Ulanov A., Boyarov K. Comparison of mechanical characteristics of vibration isolators made of wire pressed materials. International Journal of Engineering and Technology, Vol. 6, Issue 5, 2014, p. 2201-2208.
- [17] Ulanov A., And Ponomarev Yu Finite element analysis of elastic-hysteretic systems with regard to damping. Russian Aeronautics, Vol. 52, Issue 3, 2009, p. 264-270.
- [18] Yan H., Wang L., Jiang H.-Y., et al. Research on mechanical performance parameter of metal rubber. Proceedings WASE International Conference on Information Engineering, Vol. 3, 2010, p. 123-126.
- [19] Yurddas A., Ozkaya E., Boyac H. Nonlinear vibrations and stability analysis of axially moving strings having non-ideal mid-support conditions. Journal of Vibration and Control, Vol. 20, 2014, p. 518-534.
- [20] Falaleev S., Chaadaev K., Diligenskiy D. Selection of the hydrodynamic damper type for the turbomachine rotor. Life Science Journal, Vol. 11, Issue 7, 2014, p. 502-505.
- [21] Belousov A., Balyakin V., Novikov D. Experimental study of pressure distribution in lubricant layer of 'short' hydrodynamic dampers. Soviet Journal of Friction and Wear (Trenie i Iznos), Vol. 6, Issue 4, 1985, p. 56-59.
- [22] Wang S.-J., Liao M.-F. Control strategy and methods of rotor systems by an elastic support dry friction damper. Hangkong Dongli Xuebao Journal of Aerospace Power, Vol. 26, Issue 10, 2011, p. 2214-2219.
- [23] Wang S., Liao M. Experimental investigation of an active elastic support/dry friction damper on vibration control of rotor systems. International Journal of Turbo and Jet Engines, Vol. 31, Issue 1, 2014, p. 13-17.
- [24] Belousov A., Ponomarev Yu, Pronichev Yu, et al. Theory of an annular corrugated damper in the vibrator precession motion. Russian Aeronautics Vol. 52, Issue 2, 2009, p. 201-207.
- [25] Chegodaev D., And Ponomarev Yu Multilayer shock absorbers with controllable elastic hysteresis characteristics. Izvestiya Vysshikh Uchebnykh Zavedenij, Aviatsionnaya Tekhnika, Vol. 2, 1993, p. 63-67.
- [26] Liao M., Song M., Wang S. Active elastic support/dry friction damper with piezoelectric ceramic actuator. Shock and Vibration, Vol. 2014, 2014, https://doi.org/10.1155/2014/712426.
- [27] Ponomarev Y. Control of damping for multi-layer vibration insulators with dry friction on contact surfaces. International Journal of Engineering and Technology, Vol. 6, Issue 5, 2014, p. 2225-2228.
- [28] Belousov A., And Falaleev S. Gasostatic face seal with elastic working surface. Soviet Journal of Friction and Wear (Trenie i Iznos), Vol. 10, Issue 3, 1989, p. 34-38.
- [29] Gvozdev A., Melentjev V. Method of static characteristics calculation of vibroinsulators with rectilinear and circular working sections in a non-linear approach. Modern Applied Science, Vol. 9, Issue 3, 2015, p. 272-280.
- [30] Salehi M., Heshmat H., Walton J. On the frictional damping characterization of compliant bump foils. Journal of Tribology, Vol. 125, Issue 4, 2003, p. 804-813, https://doi.org/10.1115/1.1575774.
- [31] Ponomarev Y., Pronichevy, Chegodaev D., et al. Multilayer Dampers of Aircraft Engines (Mnogoslojnye Dempfery Dvigatelej Letatel'nyh Apparatov). Samara, Samara State Aerospace University, 1998, p. 234.
- [32] Ponomarev Y. On transformation of hysteresis in damper rings made of "metal rubber" pressure-tested wire material under precessional loading conditions. ARPN Journal of Engineering and Applied Sciences, Vol. 9, Issue 10, 2014, p. 1866-1872.
- [33] Hakan Y., Selçuk M., Sadettin K. Hybrid input shaping to suppress residual vibration of flexible systems. Journal of Vibration and Control, 2012, https://doi.org/10.1177/1077546311403179.
- [34] Eskin I. Transverse bend of multilayer package with dry friction on the contact surfaces by the cyclic force (Poperechnyj izgib mnogoslojnogo paketa s suhim treniem na kontaktnyh poverhnostyah ciklicheskoj siloj). Proceedings of the Strength and Reliability of Engines and Aircraft Systems at Vibration (Vibracionnaya prochnost' i nadyozhnost' dvigatelej i sistem letatel'nyh apparatov), Kujbyshevskij Aviacionnyj Institut imeni S. P. Korolyova, Vol. 30, 1967, p. 41-54.
- [35] Mualla I. H. Experimental evaluation of new friction damper device. 12th World Conference on Earthquake Engineering, 2000.

2525. CONTROL OF ELASTIC-HYSTERETIC CHARACTERISTICS OF MULTILAYER ALL-METAL VIBRATION INSULATORS WITH STRUCTURAL DAMPING. ALEXANDER S. GVOZDEV, VLADIMIR S. MELENTJEV

- [36] Takeshi S., Hideo K. Development of advanced friction sliding damper. The 14-th World Conference on Earthquake Engineering, Beijing, China, 2008.
- [37] Stefancu A., Budescu M., Paulet Crainiceanu F., et al. Numerical and experimental analysis of a passive multilayer friction damper for seismic energy dissipation friction damper for seismic energy dissipation. The 15th World Conference on Earthquake Engineering, LISBOA, 2012.
- [38] Ando Y. Lowering friction coefficient under low loads by minimizing effects of adhesion force and viscous resistance. Wear, Vol. 254, 2003, p. 965-973, https://doi.org/10.1016/S0043-1648(03)00301-6.
- [39] Ponomarev Yu, Ermakov A., Simakov O., et al. Metallic counterpart of rubber: A material for vibration and shock protection. Metal Science and Heat Treatment, Vol. 55, Issues 1-2, 2014, https://doi.org/10.1007/s11041-013-9570-3.



Alexander S. Gvozdev received candidate of technical science degree in Samara State Aerospace University, Samara, Russia, in 2010. Now he works at Department of Aircraft Engine Design. His current research interests include vibroprotection, dynamics and strength of mechanical systems, 3D-modeling.



Vladimir S. Melentjev received candidate of technical science degree in Samara State Aerospace University, Samara, Russia, in 2010. Now he works at Department of Aircraft Engine Design. His current research interests include 3D-modeling, vibroprotection, dynamics and strength of mechanical systems, rocket engines.