The meshing characteristics of planetary gear transmission system considering the effects of load and system parameters

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Received 19 January 2025; accepted 25 March 2025; published online 24 April 2025 DOI https://doi.org/10.21595/jve.2025.24800



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Abstract. In order to investigate the effects of load and system parameters on the planetary gear system, this paper develops a nonlinear time-varying dynamic model of a 12-degree-of-freedom planetary gear system. The model is employed to analyze the influence of load and system parameter variations on the meshing force, kinetic energy, strain energy, and load-sharing coefficient. The results demonstrate that the unbalance exacerbates tooth collision and increase the peak meshing force. Increasing the load will reduce the tooth collision and enhances the load sharing coefficient of the system. Decreasing the tooth side clearance, the unilateral tooth collision will be transformed into bilateral collision and mitigates the fluctuation of the meshing force. Additionally, increasing the transmission error amplifies the magnitude and fluctuation of the strain energy but reduces the system's load-sharing coefficient.

Keywords: planetary gear rotor system, nonlinear time-varying model, system meshing force, load sharing coefficient, kinetic energy and strain energy.

1. Introduction

With the development of mechanical equipment towards large-scale, high-reliability, high-accuracy and long-life, the requirements for the dynamics of planetary gear systems are also increasing. Accurately predicting the dynamic performance of these systems has become increasingly important. However, the variation of system parameters such as unbalance, tooth side clearance, variations in system parameters, such as unbalance, tooth side clearance, and transmission error, significantly increase the complexity of studying gear system dynamics. Meshing characteristics are a critical performance metric in planetary gearing systems, as they directly affect the stability, noise and lifespan of the transmission system. Since load and system parameters are key factors influencing the meshing characteristics of planetary gears, it is essential to study their effects on these characteristics.

In recent years, many scholars have carried out various researches on planetary gearing systems. Huachao Xu [1] et al. established a dynamic model of a planetary gear system with rolling bearings, analyzed the dynamic behavior of the system and the dynamic contact load of the rolling elements, and revealed the dynamic influence mechanism between the planetary gears and the rolling bearings. Tan Jianju [2] et al. used the concentrated mass method to establish a rigid-flexible coupled dynamics model of a planetary gear set, analyzed the dynamics response of the system based on the model, and summarized the effects of input torque changes on the system vibration. Siyu Wang [3] et al. developed a nonlinear dynamics model of a planetary gearbox system taking into account several nonlinear factors, and utilized the numerical integration method

via Lunger-Kutta to analyze the dynamics response when the input speed is varied. Zhang [4] et al. developed a free vibration dynamics model considering time-varying meshing stiffness to study the modal characteristics of planetary gear system, and investigated the effect of time-varying meshing stiffness on its intrinsic characteristics. Wang S. [5] et al. built a GTF gearbox systems dynamics model, the effects of the double rotor bearings on the dynamic performance of the GTF transmission system are analyzed, especially the effects of the bearing positions on the load sharing performance of the GTF gearbox star gear transmission system. Cheng Wang [6] analyzed the effect of planetary gears and star gears on the transmission efficiency by taking the closed differential double helical gear train as the object of study. Zhu L. S. [7] et al. based on the finite element method A pair of generalized axial-oscillating multi-degree-of-freedom coupled helical gears meshing dynamics model was established, and the effects of meshing stiffness and meshing damping on the intrinsic characteristics and transient response of the system were analyzed. Jie Yang [8] et al. established a coupled dynamics model of a herringbone gear rotor system supported by axial bearings, and the effect of bearing ripple degree on the system is analyzed by comparing the two system dynamics accordingly at health and failure. Zou HaoRan [9] et al. developed a coupled dynamics model of GTF planetary gearing system considering time-varying meshing stiffness and transmission error, based on which the effects of different topological modification schemes on the dynamic characteristics of the system were investigated. Ali Tatar [10] et al. assumed the gears to be rigid and the bearings to be flexible, established a finite element dynamics model of a planetary gearbox, analyzed the intrinsic properties of the system, and found that the gyroscopic effect is more significant on the planetary gearbox. Hao Dong [11] et al. established a static load sharing calculation model in order to analyze the load equalization characteristics of an aero-engine reducer, and analyzed the effects of manufacturing and installation errors on the load sharing factor. Siyu Wang [12] et al. proposed an improved nonlinear dynamics model for GTF gearboxes that includes nonlinear factors in order to make the study closer to the reality. By analyzing the dynamic characteristics, it was found that the nonlinear dynamics behavior of the system would change with the change of tooth collision state. Shuai Mo [13] et al. developed a refined new dynamic model for the vibration and intrinsic characteristic analysis of a double-helix star gear transmission system. Taking a certain type of aero-engine gear transmission system as the research object, the validity of the dynamics model is verified through calculation. Fang Guo [14] et al. developed a dynamic model of a generalized planetary gearing system containing timevarying meshing stiffness and load balance, and analyzed the effects of eccentricity error and operating frequency variation on the bifurcation characteristics of the system. Haoran Zou [15] et al. establish an improved dynamics model considering the elastic deformation of the toothed ring, and use it to compare with that without considering the elastic deformation of the toothed ring. Wei Liu [16] et al deduced the translation-torsion equations of motion of the new elastic discretization model considering the sliding friction, and investigated the effect of the sliding friction on the time-varying lattice stiffness. Dong Li [17] analyzed the distribution of loads in the planetary gear system as well as the effect of mounting errors on power distribution based on the efficiency calculation model. Lokaditya Ryali [18] et al. developed a three-dimensional dynamic model of a planetary gear system considering the effects of internal and external excitations, and analyzed the dynamic response of the system and the distribution of gear stresses under different operating conditions. Hongyuan Zhang [19] et al. established a nonlinear dynamic model of a planetary gear system with multiple degrees of freedom, and numerically solved it by the Lunger-Kutta method. The effects of nonlinear factors on the dynamic response of the system and the effects of internal and external excitations on the main frequencies of the system are analyzed. Lanlan Hou [20] et al. developed an analytical model of a nonlinear GTF planetary gear system and analyzed the nonlinear properties of the system by means of bifurcation diagrams and Poincaré diagrams, and also demonstrated that the chaotic path of the intermittent. Jungang Wang [21] et al. conducted an in-depth investigation into the evolution of load-sharing characteristics in planetary gear systems under the coupled effects of rotational speed variations and internal excitations. Yifan Huangfu [22] et al. proposed a wear prediction model for planetary gear systems. The model's prediction accuracy and reliability were systematically validated through a combined approach of finite element simulation and experimental testing. Gao Dingqiang and Gao Zhiyu [23] jointly established a nonlinear dynamic model of a multi-degree-of-freedom two-stage helical gear planetary system and systematically investigated the influence mechanisms of nonlinear factors on the system's dynamic characteristics. Li Shenlong [24] et al. constructed a multi-gear planetary gear train model by considering the positional relationships between the planetary gears and the planetary shafts, and conducted an in-depth analysis of the influence of positional errors on the system's vibration characteristics.

In summary, scholars have conducted a lot of research on planetary gear dynamics modeling and the effect of nonlinear factors on the overall vibration of the gearbox. However, studies on the influence of load and system parameters on the meshing characteristics of the planetary gear system are relatively rare. Meshing characteristics in planetary gear transmission systems significantly impact the operational stability, noise level, and lifespan of the system. Therefore. This paper establishes a nonlinear time-varying dynamics model of planetary gear system. considering coupling factors of unbalance, time-varying mesh stiffness, tooth side clearance and transmission error, and studies the effects of unbalance, load, tooth side clearance and transmission error on the meshing characteristics of the planetary gear system are investigated.

2. Planetary gear system dynamics modeling

The structure of the planetary gear rotor system is shown in Fig. 1. It primarily consists of consists of the input shaft, the sun gear (S), five identical planetary gears (pi1-pi5), ring (R), the planetary carrier, the torsion bracket, and the flexible parts connected with the torsion bracket. The sun gear is a floating part, while the ring is a semi-floating part. The planetary carrier is connected to the torsion bracket and the flexible parts, forming a cantilever beam-like structure. The sun gear, the five planetary gears and the ring are all herringbone gears. The sun gear is centrally located, and the five planetary gears are positioned at 18° , 90° , 162° , 234° , and 306° around the sun gear, respectively. After the gravity and torque generated by the meshing of the entire gearbox are transmitted to the flexible parts through the torsion bracket, the flexible parts further transfer them to the casing. The power transmission path of the system is as follows: the input shaft delivers the power to the sun gear via the spline, the sun gear then transfers the force to the planetary gears through gear meshing; the planetary gears subsequently transmit the power to the ring gear through gear meshing; finally, the ring gear, which is bolted to the output shaft, serves as the output.

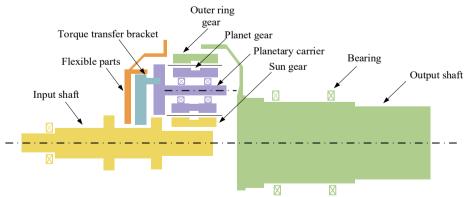


Fig. 1. Sketch of the structure of the planetary gear system

Considering the actual structure and operation effect, in order to make the analysis results more focused on the gear radial vibration, axial vibration and small swing on the overall vibration of the impact as well as to improve the efficiency and stability of the calculation, this paper uses the

following assumptions in the system modeling:

- (1) The bearing housing and other parts and foundation are rigid.
- (2) The meshing force of the gear pair always lies in the meshing plane.
- (3) Changes in gear support stiffness are ignored [25-26].

Based on the traditional numerical simulation modeling approach, the shaft segments of the rotor system are modeled using Timoshenko beam units. The left and right nodes of the beam unit are defined as Node i and Node j, respectively. Each shaft segment element has a total of 12 degrees of freedom, defined as follows:

$$u_{ij} = [x_i \quad y_i \quad z_i \quad \theta_{xi} \quad \theta_{yi} \quad \theta_{zi} \quad x_j \quad y_j \quad z_j \quad \theta_{xj} \quad \theta_{yj} \quad \theta_{zj}]^T. \tag{1}$$

The equation of motion of the shaft segment unit can be expressed as:

$$M_s \ddot{u}_{ij} + (C_s + \Omega G_s) \dot{u}_{ij} + K_s u_{ij} = F_s, \tag{2}$$

where M_s , C_s , G_s , K_s are the mass matrix, damping matrix, gyroscopic force matrix, and stiffness matrix of the axial segment unit, respectively, is the rotational speed of the axial segment unit, and is the external load of the axial segment unit.

The gears are modeled using the beam unit equivalent method. Under the premise of ensuring similarity in geometrical parameters and dynamic characteristics between the actual model and the theoretical model, the sun gear, planetary gears, and ring are simplified. The simplification method involves dividing the nodes of the beam unit at locations such as bearing support, gear meshing, and mutation of the shaft segment, and so on.

The planetary gear shaft are mounted on a fixed planetary carrier, which is connected to a flexible part, each planetary gear is supported by two angular contact ball bearings, For the convenience of numerical simulation, the planetary carrier, the flexible component, and the bearings are collectively represented as two sets of support stiffnesses [27]. The input shaft solely transmits torque to the whole system and does not provide any additional support; therefore, it is omitted in the modeling process.

Since dynamic simulation analysis of multi-gear systems is generally challenging and prone to convergence difficulties, and considering that the dynamic characteristics of the system are determined by the modal properties of each subsystem, single-axis modal analysis was adopted to validate the accuracy of the model. Single-axis modal and natural frequency analyses were performed for the planetary gear shafts and the ring shaft respectively, and the results were compared and verified using Ansys Workbench. During the analysis with Ansys Workbench, the deformation effect caused by the disc structure of the two shafts is ignored, Table 1 and Table 2 present the first two orders of natural frequencies of the two shafts obtained by the two different methods, it can be observed that the results from both methods show minimal differences. The results of the planetary gear shaft and the ring shaft are illustrated in Fig. 2 and Fig. 3, respectively. From the figures, it is evident that the first two orders natural frequencies of the planetary gear shaft and the ring shaft obtained from the numerical model are in close agreement with those obtained from Ansys Workbench. This indicates that the numerical model is highly accurate. Consequently, empirical formulas were used to calculate the gear meshing stiffness, which was then integrated with the constructed single-axis numerical model, based on this, a multi-axis finite element numerical model is further established.

Table 1. The first two orders of natural frequency of planetary gear shaft

Order	Precession direction	Anays results (Hz)	Numerical results (Hz)	Error (%)
First	Forward precession	805.2	764.0	5.09
FIISt	Backward precession	759.5	720.6	5.12
Casand	Forward precession	1317.7	1260.5	4.34
Second	Backward precession	1230.3	1177.2	4.31

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Order	Precession direction	Anays results (Hz)	Numerical results (Hz)	Error (%)							
First	Forward precession	211.6	211.7	0.05							
FIISt	Backward precession	175.0	175.5	0.27							
Casand	Forward precession	710.9	629.2	11.5							
Second	Backward precession	674.2	608.8	9.6							

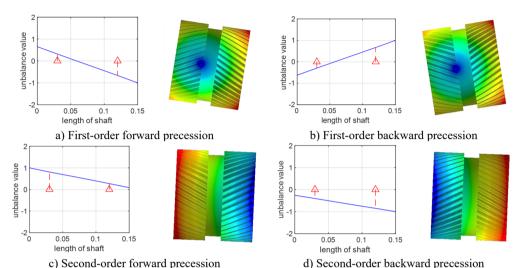


Fig. 2. First two orders of vibration patterns of planetary gear shafts

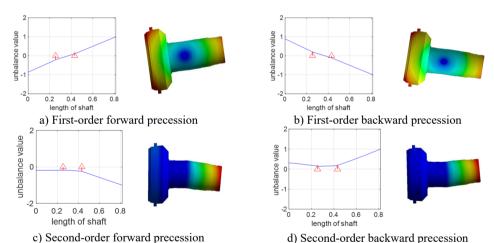
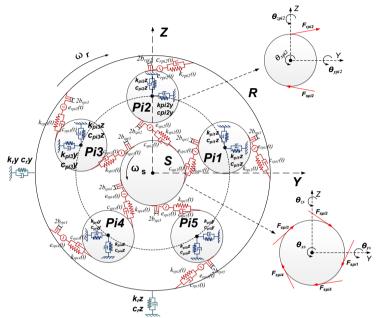


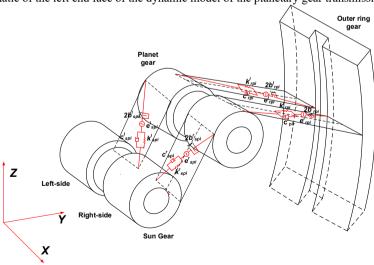
Fig. 3. First two orders of vibration patterns of ring shaft

The dynamics model of the star gear system is shown in Fig. 4, where Pi_1 , Pi_2 , Pi_3 , Pi_4 , Pi_5 represent five planetary gears. S denotes the sun gear. R represents outer gear ring, k_{spiN} , c_{spiN} (N=1,2,3,4,5) are the meshing stiffness and meshing damping between the planetary gear and the sun gear respectively. k_{spiN} and c_{spiN} (N=1,2,3,4,5) represent the meshing stiffness and meshing damping between the sun gear and the planetary gears respectively, while k_{rpiN} and c_{rpiN} (N=1,2,3,4,5) represent the meshing stiffness and meshing damping between the ring gear and the planetary gears, respectively. $k_r y$, $c_r z$, $k_{piN} y$ and $c_{piN} z$ (N=1,2,3,4,5) are the support stiffness and support damping of the ring gear and planetary gears in the horizontal and vertical directions, respectively. e_{spiN} and e_{rpiN} (N=1,2,3,4,5) represent the transmission error

between the planetary gears and the sun gear and the transmission error between the planetary gears and the ring gear. b_{spiN} and b_{rpiN} (N=1, 2, 3, 4, 5) denote the unilateral tooth side clearance between the planetary gears and the sun gear and between the planetary gears and the ring gear, respectively. F_{spiN} and F_{rpiN} (N=1, 2, 3, 4, 5) represent the external meshing force between the planetary gear and the sun gear and the internal meshing force between the planetary gear and the ring gear, respectively. The superscripts l and r in Fig. 4(b) indicate the left-side gear and the right-side gear, respectively.



a) Schematic of the left end face of the dynamic model of the planetary gear transmission system



b) 3D schematic diagram of the dynamics model of the planetary gear transmission system **Fig. 4.** Dynamic model planetary gear transmission system

During gear meshing, the engagement of gear teeth alternates, causing the number of teeth simultaneously in contact to vary periodically over time. Additionally, the alternation between

single and double tooth meshing leads to periodic changes in the overall meshing stiffness of helical gears, thereby inducing dynamic stiffness fluctuations. Consequently, the stiffness characteristics of the mesh exhibit distinct cyclic patterns during transmission, which can be effectively represented using a Fourier series [28], expressed as:

$$k(t) = k_0 + \sum_{j=1}^{\infty} \left[a_j \cos(j\omega_{\Lambda}t) + b_j \sin(j\omega_{\Lambda}t) \right] = k_0 + \sum_{j=1}^{\infty} k_j \cos(j\omega_{\Lambda}t + \varphi_j), \qquad (3)$$

where k_0 is the average meshing stiffness a_i , b_i , k_i are the Fourier series, φ_j is the phase angle, ω_A is the meshing frequency, and is expressed as:

$$\omega_{\Lambda} = \frac{\pi n_i Z_i}{30},\tag{4}$$

where, n_i is the gear speed and Z_i is the number of gear teeth.

Since the system is more sensitive to the dynamic characteristics in the low frequency range, the effect of higher order terms is usually smaller, and to improve the computational efficiency, the Fourier series of the second order and above is ignored, it can be written as:

$$k(t) = k_0 [1 + \varepsilon \sin(\omega_{\Lambda} t + \varphi)], \tag{5}$$

where ε is the stiffness fluctuation amplitude.

Gear transmission error primarily arises from inaccuracies in design, manufacturing, and assembly, leading to positional deviations during gear meshing that result in tooth impacts. This error exhibits periodicity and can be accurately modeled using a Fourier series formulation [29], as following:

$$e(t) = e_0 + \sum_{j=1}^{\infty} \left[c_j \cos(j\omega t) + d_j \sin(j\omega t) \right] = e_0 + \sum_{j=1}^{\infty} e_j \cos(j\omega_{\Lambda} t + \varphi_j), \tag{6}$$

where e_0 is the average transmission error c_i , d_i is the Fourier series.

Similarly, if the Fourier series of second order and higher are neglected and the mean error $e_0 = 0$, it can be written as:

$$e(t) = e_{\rm s}\sin(\omega_{\rm A}t + \varphi),\tag{7}$$

where, e_s is the amplitude of transmission error fluctuations

Due to the need for lubrication, as well as gear manufacturing and installation errors, mutual meshing between the teeth of the wheel there is a tooth side clearance. A tooth side clearance exists between the meshing gear teeth. The presence of tooth side clearance alters the contact state between the gear teeth, leading to continuous impacts that significantly affect the dynamic characteristics of the gear system, The gear deflection function can be expressed as [30]:

$$g(\Delta) = \begin{cases} \Delta - b_n, & \Delta > b_n, \\ 0, & -b_n \le \Delta \le b_n, \\ \Delta + b_n, & \Delta < -b_n, \end{cases}$$
(8)

where b_n is the unilateral tooth side clearance, Δ is the relative meshing line length.

3. Planetary gear system dynamics modeling

All gears in the system are herringbone gears, with their geometric parameters listed in Table 3

and operational parameters detailed in Table 4. The meshing of herringbone gears can be likened to two helical gears engaging simultaneously but rotating in opposite directions [31]. As a result, the dynamic equations for one side can be derived from those of the other. Due to space constraints, this paper focuses solely on deriving the dynamic equations for the left-side gears of the planetary gear system. Based on Fig. 4(a), the dynamic equations for the planetary gear pi2 and the sun gear are obtained as follows:

Planetary gear pi2:

$$\begin{cases} m_{pi2}\ddot{x}_{pi2} + c_{pi2x}\dot{x}_{pi2} + k_{pi2x}x_{pi2x} + \left[c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)\right] \sin\beta \\ -\left[c_{rpi2}\dot{p}_{rpi2} + k_{rpi1}g(t)\right] \sin\beta = 0, \\ m_{pi1}\ddot{y}_{pi2} + c_{pi2y}\dot{y}_{pi2} + k_{pi2y}y_{pi2y} + \left[c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)\right] \cos\beta \sin\psi_{ij} \\ +\left[c_{rpi2}\dot{p}_{rpi2} + k_{rpi2}g(t)\right] \cos\beta \sin\psi_{ij} = 0, \\ m_{pi2}\ddot{z}_{pi2} + c_{pi2z}\dot{z}_{pi2} + k_{pi2z}z_{pi2z} - \left[c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)\right] \cos\beta \cos\psi_{ij} \\ +\left[c_{rpi2}\dot{p}_{rpi2} + k_{rpi2}g(t)\right] \cos\beta \cos\psi_{ij} = 0, \\ I_{ppi2}\ddot{\theta}_{xpi2} + c_{pi2\theta x}\dot{\theta}_{xpi2} + r_{pi2}\left[c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)\right] \cos\beta \\ -r_{pi2}\left[c_{rpi2}\dot{p}_{rpi2} + k_{rpi2}g(t)\right] \cos\beta = 0, \\ I_{ppi2}\ddot{\theta}_{ypi2} + c_{pi2\theta y}\dot{\theta}_{ypi2} + k_{pi2\theta y}\theta_{ypi2} + r_{pi2}\left[c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)\right] \sin\beta \sin\psi_{ij} \\ +r_{pi2}\left[c_{rpi2}\dot{p}_{rpi2} + k_{rpi2}g(t)\right] \sin\beta \sin\psi_{ij} = 0, \\ I_{ppi2}\ddot{\theta}_{zpi2} + c_{pi2\theta z}\dot{\theta}_{zpi2} + k_{pi2\theta z}\theta_{zpi2} + r_{pi2}\left[c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)\right] \sin\beta \cos\psi_{ij} \\ -r_{pi2}\left[c_{rpi2}\dot{p}_{rpi2} + k_{rpi2}g(t)\right] \sin\beta \cos\psi_{ij} = 0. \end{cases}$$

Sun gear:

$$\begin{cases} m_{s}\ddot{x}_{s} + c_{s}\dot{x}_{s} + k_{sx}x_{sx} - \sum_{n=1}^{5} [c_{spin}\dot{p}_{spin} + k_{spin}g(t)]\sin\beta = 0 \\ m_{s}\ddot{y}_{s} + c_{s}\dot{y}_{s} + k_{sy}y_{sy} + [c_{spi1}\dot{p}_{spi1} + k_{spi1}g(t)]\sin\beta + [c_{spi4}\dot{p}_{spi4} + k_{spi4}g(t)]\sin\beta \\ + [c_{spi5}\dot{p}_{spi5} + k_{spi5}g(t)]\sin\beta - [c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)]\sin\beta \\ - [c_{spi3}\dot{p}_{spi3} + k_{spi3}g(t)]\sin\beta = 0, \\ m_{s}\ddot{z}_{s} + c_{s}\dot{z}_{s} + k_{sz}z_{sz} + \sum_{n=1}^{2} [c_{spin}\dot{p}_{spin} + k_{spin}g(t)]\cos\beta \\ - \sum_{n=3}^{5} [c_{spin}\dot{p}_{spin} + k_{spin}g(t)]\cos\beta = T_{i}, \\ I_{ps}\ddot{\theta}_{xs} + c_{s\theta x}\dot{\theta}_{xs} + [c_{spi1}\dot{p}_{spi1} + k_{spi1}g(t)]\sin\beta\sin\psi_{ij} \\ - [c_{spi2}\dot{p}_{spi2} + k_{spi2}g(t)]\sin\beta\sin\psi_{ij} + \sum_{n=3}^{5} [c_{spin}\dot{p}_{spin} + k_{spin}g(t)]\sin\beta\sin\psi_{ij} = 0, \\ I_{ps}\ddot{\theta}_{ys} + c_{s\theta y}\dot{\theta}_{ys} + k_{s\theta y}\theta_{ys} - [c_{spi1}\dot{p}_{spi1} + k_{spi1}g(t)]\sin\beta\cos\psi_{ij} \\ - [c_{spi5}\dot{p}_{spi5} + k_{spi5}g(t)]\sin\beta\cos\psi_{ij}, \\ + \sum_{n=2}^{4} [c_{spin}\dot{p}_{spin} + k_{spin}g(t)]\sin\beta\cos\psi_{ij} + k_{spi1}g(t)]\sin\beta\cos\psi_{ij} \\ - [c_{spi5}\dot{p}_{spi5} + k_{s\theta z}\theta_{zs} - [c_{spi1}\dot{p}_{spi1} + k_{spi1}g(t)]\sin\beta\cos\psi_{ij} \\ - [c_{spi5}\dot{p}_{spi5} + k_{spi5}g(t)]\sin\beta\cos\psi_{ij} + \sum_{n=2}^{4} [c_{spin}\dot{p}_{spin} + k_{spin}g(t)]\sin\beta\cos\psi_{ij} = 0. \end{cases}$$

The dynamic equations for the other gears are similar, and thus can be expressed as: Planetary gear pi1:

$$\begin{cases} m_{pi1} \ddot{x}_{pi1} + c_{pi1x} \dot{x}_{pi1} + k_{pi1x} x_{pi1x} + [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \sin\beta \\ -[c_{rpi1} \dot{p}_{rpi1} + k_{rpi1} g(t)] \sin\beta = 0, \\ m_{pi1} \ddot{y}_{pi1} + c_{pi1y} \dot{y}_{pi1} + k_{pi1y} y_{pi1y} - [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \cos\beta \sin\psi_{ij} \\ +[c_{rpi1} \dot{p}_{rpi1} + k_{rpi1} g(t)] \cos\beta \sin\psi_{ij} = 0, \\ m_{pi1} \ddot{z}_{pi1} + c_{pi1z} \dot{z}_{pi1} + k_{pi1z} z_{pi1z} - [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \cos\beta \cos\psi_{ij} \\ -[c_{rpi1} \dot{p}_{rpi1} + k_{rpi1} g(t)] \cos\beta \cos\psi_{ij} = 0, \\ I_{ppi1} \ddot{\theta}_{xpi1} + c_{pi1\theta x} \dot{\theta}_{xpi1} + r_{pi1} [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \cos\beta \\ -r_{pi1} [c_{rpi1} \dot{p}_{rpi1} + k_{rpi1} g(t)] \cos\beta = 0, \\ I_{ppi1} \ddot{\theta}_{ypi1} + c_{pi1\theta y} \dot{\theta}_{ypi1} + k_{pi1\theta y} \theta_{ypi1} + r_{pi1} [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \sin\beta \sin\psi_{ij} \\ +r_{pi1} [c_{rpi1} \dot{p}_{rpi1} + k_{rpi1} g(t)] \sin\beta \sin\psi_{ij} = 0, \\ I_{ppi1} \ddot{\theta}_{zpi1} + c_{pi1\theta z} \dot{\theta}_{zpi1} + k_{pi1\theta z} \theta_{zpi1} + r_{pi1} [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \sin\beta \cos\psi_{ij} \\ +r_{pi1} [c_{rpi1} \dot{p}_{rpi1} + k_{rpi1} g(t)] \sin\beta \cos\psi_{ij} = 0. \end{cases}$$

Planetary gear pi3:

$$\begin{pmatrix} m_{pi3}\ddot{x}_{pi3} + c_{pi3}\dot{x}_{pi3} + k_{pi3x}x_{pi3x} + [c_{spi3}\dot{p}_{spi3} + k_{spi3}g(t)]\sin\beta \\ -[c_{rpi3}\dot{p}_{rpi3} + k_{rpi3}g(t)]\sin\beta = 0, \\ m_{pi3}\ddot{y}_{pi3} + c_{pi3}\dot{y}_{pi3} + k_{pi3y}y_{pi3y} + [c_{spi3}\dot{p}_{spi3} + k_{spi3}g(t)]\cos\beta\sin\psi_{ij} \\ -[c_{rpi3}\dot{p}_{rpi3} + k_{rpi3}g(t)]\cos\beta\sin\psi_{ij} = 0, \\ m_{pi3}\ddot{z}_{pi3} + c_{pi3}\dot{z}_{pi3} + k_{pi3z}z_{pi3z} + [c_{spi3}\dot{p}_{spi3} + k_{spi3}g(t)]\cos\beta\cos\psi_{ij} \\ -[c_{rpi3}\dot{p}_{rpi3} + k_{rpi3}g(t)]\cos\beta\cos\psi_{ij} = 0, \\ I_{ppi3}\ddot{\theta}_{xpi3} + c_{pi3\theta x}\dot{\theta}_{xpi3} + r_{pi3}[c_{spi3}\dot{p}_{spi3} + k_{spi3}g(t)]\cos\beta \\ -r_{pi3}[c_{rpi3}\dot{p}_{rpi3} + k_{rpi3}g(t)]\cos\beta = 0, \\ I_{ppi3}\ddot{\theta}_{ypi3} + c_{pi3\theta y}\dot{\theta}_{ypi3} + k_{pi3\theta y}\theta_{ypi3} + r_{pi3}[c_{spi3}\dot{p}_{spi3} + k_{spi3}g(t)]\sin\beta\sin\psi_{ij} \\ +r_{pi3}[c_{rpi3}\dot{p}_{rpi3} + k_{rpi3}g(t)]\sin\beta\sin\psi_{ij} = 0, \\ I_{ppi3}\ddot{\theta}_{zpi3} + c_{pi3\theta z}\dot{\theta}_{zpi3} + k_{pi3\theta z}\theta_{zpi3} - r_{pi3}[c_{spi3}\dot{p}_{spi3} + k_{spi3}g(t)]\sin\beta\cos\psi_{ij} \\ -r_{pi2}[c_{rpi3}\dot{p}_{rpi3} + k_{rpi3}g(t)]\sin\beta\cos\psi_{ij} = 0. \end{cases}$$

Planetary gear pi4:

$$m_{pi4}\ddot{x}_{pi4} + c_{pi4}\dot{x}_{pi4} + k_{pi4x}x_{pi4x} + [c_{spi4}\dot{p}_{spi4} + k_{spi4}g(t)]\sin\beta$$

$$-[c_{rpi4}\dot{p}_{rpi4} + k_{rpi4}g(t)]\sin\beta = 0,$$

$$m_{pi4}\ddot{y}_{pi4} + c_{pi4}\dot{y}_{pi4} + k_{pi4y}y_{pi4y} - [c_{spi3}\dot{p}_{spi4} + k_{spi4}g(t)]\cos\beta\sin\psi_{ij}$$

$$-[c_{rpi4}\dot{p}_{rpi4} + k_{rpi4}g(t)]\cos\beta\sin\psi_{ij} = 0,$$

$$m_{pi4}\ddot{z}_{pi4} + c_{pi4}\dot{z}_{pi4} + k_{pi4z}z_{pi4z} + [c_{spi4}\dot{p}_{spi4} + k_{spi4}g(t)]\cos\beta\cos\psi_{ij}$$

$$+[c_{rpi4}\dot{p}_{rpi4} + k_{rpi4}g(t)]\cos\beta\cos\psi_{ij} = 0,$$

$$I_{ppi4}\ddot{\theta}_{xpi4} + c_{pi4\thetax}\dot{\theta}_{pi4} + r_{pi4}[c_{spi4}\dot{p}_{spi4} + k_{spi4}g(t)]\cos\beta$$

$$-r_{pi4}[c_{rpi4}\dot{p}_{rpi4} + k_{rpi4}g(t)]\cos\beta = 0,$$

$$I_{ppi4}\ddot{\theta}_{ypi4} + c_{pi4\thetay}\dot{\theta}_{ypi4} + k_{pi4\thetay}\theta_{ypi4} - r_{pi4}[c_{spi4}\dot{p}_{spi4} + k_{spi4}g(t)]\sin\beta\sin\psi_{ij}$$

$$+r_{pi4}[c_{rpi4}\dot{p}_{rpi4} + k_{rpi4}g(t)]\sin\beta\sin\psi_{ij} = 0,$$

$$I_{ppi4}\ddot{\theta}_{zpi4} + c_{pi4\thetaz}\dot{\theta}_{zpi4} + k_{pi4\thetaz}\theta_{zpi4} - r_{pi4}[c_{spi4}\dot{p}_{spi4} + k_{spi4}g(t)]\sin\beta\cos\psi_{ij}$$

$$+r_{pi4}[c_{rpi4}\dot{p}_{rpi4} + k_{rpi4}g(t)]\sin\beta\cos\psi_{ij} = 0.$$

$$I_{ppi4}\ddot{\theta}_{zpi4} + c_{pi4\thetaz}\dot{\theta}_{zpi4} + k_{pi4\thetaz}\theta_{zpi4} - r_{pi4}[c_{spi4}\dot{p}_{spi4} + k_{spi4}g(t)]\sin\beta\cos\psi_{ij}$$

$$+r_{pi4}[c_{rpi4}\dot{p}_{rpi4} + k_{rpi4}g(t)]\sin\beta\cos\psi_{ij} = 0.$$

Planetary gear pi5:

$$\begin{cases} m_{pi5} \ddot{x}_{pi5} + c_{pi5} \dot{x}_{pi5} + k_{pi5x} x_{pi5x} + [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \sin\beta \\ -[c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \sin\beta = 0, \\ m_{pi5} \ddot{y}_{pi5} + c_{pi5} \dot{y}_{pi5} + k_{pi5y} y_{pi5y} - [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \cos\beta \sin\psi_{ij} \\ -[c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \cos\beta \sin\psi_{ij} = 0, \\ m_{pi5} \ddot{z}_{pi5} + c_{pi5} \dot{z}_{pi5} + k_{pi5z} z_{pi5z} + [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \cos\beta \cos\psi_{ij} \\ -[c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \cos\beta \cos\psi_{ij} = 0, \\ I_{ppi5} \ddot{\theta}_{xpi5} + c_{pi5\theta x} \dot{\theta}_{xpi5} + r_{pi5} [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \cos\beta \\ -r_{pi5} [c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \cos\beta = 0, \\ I_{ppi5} \ddot{\theta}_{ypi5} + c_{pi5\theta y} \dot{\theta}_{ypi5} + k_{pi5\theta y} \theta_{ypi5} + r_{pi5} [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \sin\beta \sin\psi_{ij} \\ -r_{pi5} [c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \sin\beta \sin\psi_{ij} = 0, \\ I_{ppi5} \ddot{\theta}_{zpi5} + c_{pi5\theta z} \dot{\theta}_{zpi5} + k_{pi5\theta z} \theta_{zpi5} + r_{pi5} [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \sin\beta \cos\psi_{ij} \\ -r_{pi4} [c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \sin\beta \cos\psi_{ij} = 0. \end{cases}$$

Ring:

$$\begin{cases} m_r \ddot{x}_r + c_r \dot{x}_r + k_{rx} x_{rx} + \sum_{n=1}^5 [c_{rpin} \dot{p}_{rpin} + k_{rpin} g(t)] \sin\beta = 0, \\ m_r \ddot{y}_r + c_r \dot{y}_r + k_{ry} y_{ry} - \sum_{n=1}^3 [c_{rpin} \dot{p}_{rpin} + k_{rpin} g(t)] \sin\beta \\ + [c_{rpi4} \dot{p}_{rpi4} + k_{rpi4} g(t)] \sin\beta + [c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \sin\beta = 0, \\ m_s \ddot{z}_s + c_s \dot{z}_s + k_{sz} z_{sz} + [c_{rpi1} \dot{p}_{rpi1} + k_{rpi1} g(t)] \sin\beta + [c_{rpi5} \dot{p}_{rpi5} + k_{rpi5} g(t)] \sin\beta \\ - \sum_{n=2}^4 [c_{spin} \dot{p}_{spin} + k_{spin} g(t)] \cos\beta \sin\psi_{ij} = 0, \\ l_{pr} \ddot{\theta}_{xr} + c_{r\theta x} \dot{\theta}_{xr} + \sum_{n=1}^5 [c_{spin} \dot{p}_{spin} + k_{spin} g(t)] \sin\beta = -T_t, \\ l_{pr} \ddot{\theta}_{yr} + c_{r\theta x} \dot{\theta}_{yr} + k_{r\theta y} \theta_{ys} - [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \sin\beta \sin\psi_{ij} \\ - [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \sin\beta \sin\psi_{ij} \\ + \sum_{n=2}^4 [c_{spin} \dot{p}_{spin} + k_{spin} g(t)] \sin\beta \sin\psi_{ij} = 0, \\ l_{pr} \ddot{\theta}_{zr} + c_{r\theta z} \dot{\theta}_{zr} + k_{r\theta z} \theta_{zr} - [c_{spi1} \dot{p}_{spi1} + k_{spi1} g(t)] \sin\beta \cos\psi_{ij} \\ - [c_{spi2} \dot{p}_{spi2} + k_{spi2} g(t)] \sin\beta \cos\psi_{ij} - [c_{spi4} \dot{p}_{spi4} + k_{spi4} g(t)] \sin\beta \cos\psi_{ij} \\ + [c_{spi3} \dot{p}_{spi3} + k_{spi3} g(t)] \sin\beta \cos\psi_{ij} + [c_{spi5} \dot{p}_{spi5} + k_{spi5} g(t)] \sin\beta \cos\psi_{ij} = 0, \end{cases}$$

where k_{ix} , k_{iy} , k_{iz} , $k_{i\theta y}$, $k_{i\theta z}$ ($i \in [pi1, pi2, pi3, pi4, pi5, s, r]$) are the stiffness of the gear i in the x, y, z direction and around the y, z axis, respectively, and c_{ix} , c_{iy} , c_{iz} , $c_{i\theta y}$, $c_{i\theta z}$ ($i \in [pi1, pi2, pi3, pi4, pi5, s, r]$) are the damping of the gear in the x, y, z direction and around the y, z axis, respectively, k_{ij} , c_{ij} ($i \in [pi1, pi2, pi3, pi4, pi5, s, r]$, $j \in [s, r]$) are the meshing stiffness and meshing damping of gears i and gears j, respectively, β is the helix angle ($\beta > 0$ if the gears are rotated right, then; $\beta < 0$ if the gears are rotated left, then), T_i and T_t are the input torque moment and system load, g(t) is the gap function, which is denoted by:

$$g_{ij}(t) = \begin{cases} p_{ij} - b_{ij}, & p_{ij} > b_{ij}, \\ 0, & |p_{ij}| \le b_{ij}, \\ p_{ij} + b_{ij}, & p_{ij} < -b_{ij}, \end{cases}$$
(16)

where b_{ij} is the unilateral tooth clearance between gear i and gear j.

 ψ_{ij} is the directional angle from the positive direction of the z-axis to the meshing surface, which is related to the direction of rotation of the driving gear, let α_{ij} denote the pressure angle and ϕ_{ij} is the direction angle of the gear. Thus, ψ_{ij} can be expressed as.

When the gear pair is an external meshing pair:

$$\psi_{ij} = \begin{cases} \alpha_{ij} - \phi_{ij}, & \text{turn clockwise,} \\ -\alpha_{ij} - \phi_{ij}, & \text{turn counterclockwise.} \end{cases}$$
 (17)

When the gear pair is an internal mesh pair, ψ_{ij} is expressed as:

$$\psi_{ij} = \begin{cases} \alpha_{ij} + \phi_{ij}, & \text{turn clockwise,} \\ \pi - \alpha_{ij} - \phi_{ij}, & \text{turn counterclockwise.} \end{cases}$$
 (18)

Each gear has three translational degrees of freedom x, y, z and three rotational degrees of freedom θ_x , θ_y , θ_z , Thus the planetary gear system has 42 degrees of freedom, can be represented as follows:

$$X = [x_1 \quad y_1 \quad z_1 \quad \theta_{x1} \quad \theta_{y1} \quad \theta_{z1} \quad \dots \quad x_7 \quad y_7 \quad z_7 \quad \theta_{x7} \quad \theta_{y7} \quad \theta_{z7}]^T. \tag{19}$$

Therefore, Eqs. (9-15) can be written as:

$$M\ddot{X} + (C_s + C_m)\dot{X} + K_sX + K_mG(p) = F,$$
 (20)

where M is the mass matrix of the gear system, C_s is the support damping matrix of the gear system, K_s is the support stiffness matrix of the gear system, C_m is the meshing damping matrix of the gear system, C_m is the meshing stiffness matrix of the gears, G(p) is the backlash function with respect to p, F is the external load.

Table 3. Gear geometric parameters

	Sun gear	Planetary gear	Ring			
Number of teeth	48	122				
Modulus (mm)	3.5					
Pressure angle (°)	22.5					
Helix angle (°)	20					

Table 4. Operating parameters of the planetary gear system

- 1 - 81	1 10
Working speed (r/min)	Rated load (N·m)
9000	9840

4. Mesh characterization of star gear system

In the actual processing and manufacturing process, it is challenging to completely eliminate unbalance within the planetary gear system. Additionally, due to the complex and unique nonlinear meshing characteristics of the planetary gear system, the meshing force exhibit highly complex nonlinear behavior. The unbalanced vibration of the rotor system significantly affects these characteristics.

The formula for the permissible unbalance amount of the shaft segment is:

$$M = m \cdot e_{ner},\tag{21}$$

where M is the unbalance amount, m is the mass of the shaft segment, e_{per} is the permissible

unbalance rate, which can be calculated using the following formula:

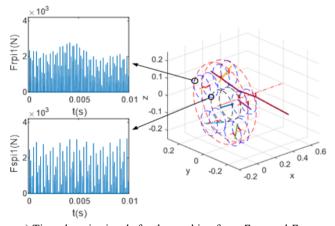
$$e_{per} = \frac{G}{1000\Omega'} \tag{22}$$

where, G represents the rotor balance quality, typically G6.3 for GTF rotors, and Ω is the maximum operating angular velocity of the rotor.

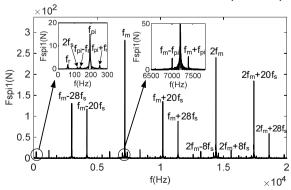
To investigate the impact of unbalance on the meshing force, it is assumed that the unbalance is increased to 10 times the permissible unbalance and exists only on the five planetary gears and the ring gear. Using unbalance loading case 1 in Table 5 (where ' \circ ' indicates inclusion and ' \times ' indicates exclusion) as an example, the external meshing force between the planetary gears and the sun gear, as well as the internal meshing force between the planetary gears and the ring gear, are analyzed. Under the light load condition of operating speed and 1 % rated load, the nonlinear time-varying model of the planetary gear system proposed in this paper is utilized to obtain the time-domain signals and frequency-domain signals of F_{spi1} and F_{rpi1} respectively, The results are shown in Fig. 5.

Table 5. Degree of unbalance loading method

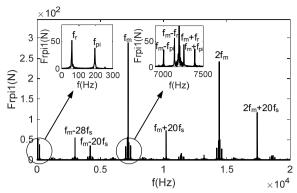
	Pi1	Pi2	Pi3	Pi4	Pi5	R
Degree of unbalance (g.mm)	750	750	750	750	750	5011
Initial phase (deg)	0	0	0	0	0	0
Case 1	0	0	0	0	0	0
Case 2	0	×	×	×	×	×



a) Time-domain signals for the meshing force F_{spi1} and F_{rpi1}



b) Frequency-domain signal of the meshing force F_{spi1}



c) Frequency-domain signal of the meshing force F_{rpi1}

Fig. 5. Time and frequency domain plots of meshing force for degree of unbalance loading case 1

Fig. 5(a) shows the time-domain plots of the meshing forces F_{spi1} and F_{rpi1} . It can be observed that under the influence of multiple unbalanced excitations, the meshing forces F_{sni1} and F_{rni1} exhibit significant collisions with a certain periodicity. Fig. 5(b) shows the frequency-domain plots of the meshing force F_{spi1} . From the figure, it is evident that the frequency-domain signal of the meshing force F_{spi1} mainly consists of the unbalanced excitation f_{pi} (194.6 Hz), f_r (59.1 Hz) and its multiplicative frequency, the meshing frequency f_m (7200 Hz) and its multiplicative frequency, as well as some side-band frequencies, for example, near f_m , there are frequencies that differ from it by integer multiples of the sun gear rotational frequency f_s (150 Hz). Additionally, in the low-frequency region, frequencies that differ from f_{pi} by integer multiples of f_r are observed. Fig. 5(c) shows the frequency-domain plot of the meshing force F_{rpi1} , whose frequencydomain signal is similar to that of the meshing force F_{spi1} . The same unbalanced excitations f_{pi} and f_{pi} appear in the low-frequency region. In the high-frequency region, besides the meshing frequency and the side-band frequencies centered around the meshing frequency with intervals of integer multiples of f_s and f_{vi} , side-band frequencies centered around the meshing frequency with intervals of f_r also present. By analyzing the side-band frequency intervals of the frequencydomain signals in Figs. 5(b) and 5(c), the modulation mechanism of the meshing side-band frequency in the planetary gear system under multi-axis unbalance can be deduced. Specifically, the meshing force is first modulated by the rotational frequency f_s of the sun gear, then by the unbalanced excitation f_{pi} of the planetary gears, and finally by the unbalanced excitation f_r of the ring gear. Therefore, the side-band frequency f_{t3} , which results from three frequency modulations centered on the meshing frequency f_m and its octave, can be expressed as:

$$f_{t3} = nf_m \pm m_1 f_s \pm m_2 f_{pi} \pm m_3 f_r, \tag{23}$$

where m_1 , m_2 , m_3 are non-negative integers.

To investigate the impact of uniaxial imbalance on both internal and external meshing forces, this paper employed unbalance loading mode 2 from Table 3. Specifically, the examined the external meshing force F_{spi1} between the planetary gear Pi1 and the sun gear, as well as the internal meshing force F_{spi1} within the ring gear, were examined. Using the nonlinear time-varying model of the planetary gear system developed in this study, time-domain and frequency-domain signals of the system were obtained at operational speed and under a light load of 1 % of the rated capacity. These results are depicted in Fig. 6.

Fig. 6 shows the time and frequency domain plots of the meshing force under the unbalance loading mode 2, where only the planetary gear Pi1 exhibits unbalance. Specifically, when the loading mode 2 is adopted, the collision between the gear teeth of the external meshing force F_{svi1}

is more intense compared to that for internal meshing force F_{rpi1} . However, overall, the meshing forces exhibit a more uniform variation compared to the case of multi-axis unbalance. The frequency-signal primarily consists of the excitation frequency f_{pi} due to the unbalance of the planetary gear Pi1, the mesh frequency f_m and its multiplicative frequency, as well as some sideband frequencies.

Comparing Fig. 5(a) and Fig. 6(a), it can be found that the amplitude of F_{spi1} increases due to the absence of the unbalance excitation from the other axes, while the collision between the planetary gear and the sun gear becomes more intense. This indicates that the unbalance vibration of each axis not only exacerbates the collision between the teeth of the gears on that axis, increasing the peak meshing load, but also affects the other axes. Due to the absence of unbalance, the amplitude of F_{rni1} is slightly decreases, the collision between the gear teeth is reduced, and the meshing force becomes more uniform. From Figs. 6(b) and (c), it can be seen that when the planetary gear exhibit unbalance, the frequency-domain signals of the internal and external meshing forces do not significantly. Compared with Figs. 5(b) and (c), the unbalance excitation of the ring gear disappears in the low-frequency region, the high-frequency region, the sideband frequencies centered around the meshing frequency are integer multiples of the meshing frequency. Specifically, the sideband frequency with an integer multiple of f_r disappears, while the amplitude of the side-band frequency with integer multiples of f_s and f_{pi} centered on the meshing frequency increases significantly. Therefore, it can evident that the magnitude of the side-band frequency is also closely related to the collision between the gear teeth. Similarly, analyzing the intervals between the side-band frequencies, the modulation mechanism of the meshing side-band frequencies in the planetary gear system under single-axis unbalance can be deduced. The meshing force is firstly modulated by the rotational frequency of the sun gear and then by the excitation frequency of the unbalance. The first modulated frequencies f_{t1} centered on the n-fold frequency of f_m can be expressed as:

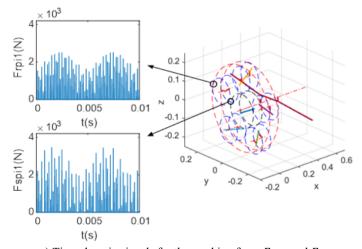
$$f_{t1} = nf_m \pm mf_s, \tag{24}$$

where m is a natural number.

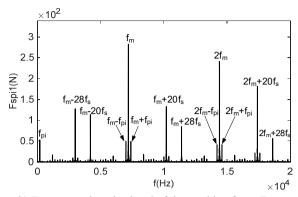
The frequency f_{t2} of the second modulation centered on f_{t1} can be expressed as:

$$f_{t2} = f_{t1} \pm m f_i, \tag{25}$$

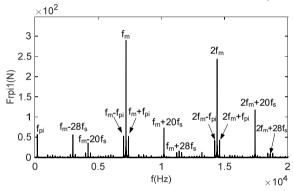
where m is a natural number, i is either Pi or R.



a) Time-domain signals for the meshing force F_{rpi1} and F_{spi1}



b) Frequency-domain signal of the meshing force F_{spi1}



c) Frequency-domain signal of the meshing force F_{rpi1}

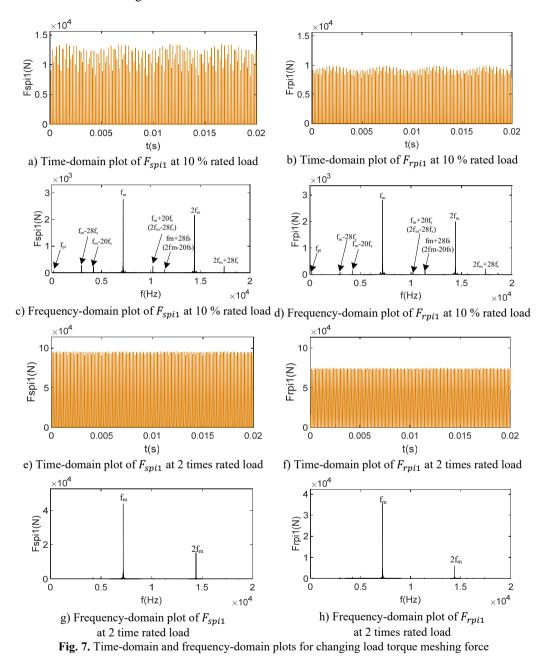
Fig. 6. Time and frequency domain plots of the meshing force for degree of unbalance loading case 2

To more clearly compare the change of gear meshing force under the influence of load, only for the unbalance loading mode 2 shown in Table 3 is analyzed. The load torque is adjusted to 10 % of the rated load and 2 times the rated load, respectively, to obtain the time-domain and frequency-domain signals of the meshing forces F_{spi1} and F_{rpi1} , as shown in Fig. 7.

In Fig. 7, when the load is increased from 1 % of the rated load to 10 % of the rated load, The average meshing forces increase significantly, and the collisions still occur. However, compared to Fig. 6, the collision phenomenon is reduced, and the variation of the meshing force within the meshing period is more uniform. From Fig. 7(c) and Fig. 7(d), it can be observed that the frequency-domain signals are still dominated by the meshing frequency f_m and its octave. Additionally, the frequencies resulting from the first and second modulations of the meshing frequency f_m are present. However, compared with Fig. 6(b) and Fig. 6(c), the amplitude of the side-band frequencies is significantly reduced, and their magnitudes are much smaller than the center frequency. This indicates that the load has a certain inhibitory effect on the side-band frequencies. This is because the increase load prolongs continuous meshing time between the gear teeth, and as the load increases, collisions between the meshing tooth surfaces disappear. From Fig. 7(e) and Fig. 7(f), it can be seen that when the system load increases to 2 times the rated load, collisions between the gears disappears, and the gears enter into a continuous meshing state. The gear meshing force is further increases, and the meshing uniformity improves. The modulated side-band frequency disappears, and the unbalance excitation becomes much smaller than the mesh frequency and is negligible. This demonstrates that the collision between the gears are one of the main reasons for the generation of the side-band frequencies in the frequency-domain

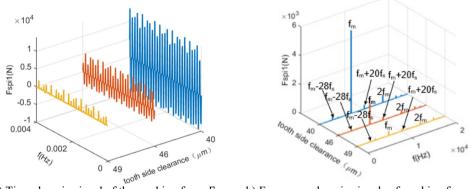
For the unbalance loading mode 1, the output load is set to be 1 % of the rated load, and the

tooth side clearance of the gear pair is changed to 49 μ m, 46 μ m and 20 μ m, respectively, and The results are shown in Fig. 8.



When the tooth side clearance is 200 µm, the meshing gears only have contact on one side of the tooth surface during the meshing process, while the other side remains free. Under reverse torque and vibration, the gears disengage, resulting in unilateral collision. Consequently, the meshing gears are in a unilateral reciprocating collision state of contact-meshing-bouncing. Within a certain range, the tooth side clearance does not significantly affect the meshing characteristics. However, when the tooth side clearance is reduced to 49 µm, as shown in Fig. 8,

the displacement amplitude of the meshing teeth between the planetary gear and the sun gear under dynamic load may exceed the remaining clearance due to the reduced tooth side clearance. This causing the other side of the tooth surface to come into contact. As both sides of the tooth surface are alternately loaded, the collision transitions from unilateral to bilateral, leading to an increase in the peak meshing force and more pronounced periodicity. In the frequency-domain signal, the peak at the meshing frequency increases, while the peaks at other side-band frequencies show little change. When the tooth side clearance is further reduced to 46 μ m, the meshing gear pair completely transitions from unilateral to bilateral collision. In the frequency-domain signal, the peak at the meshing frequency increases, while the peaks at other side-band frequencies remain largely unchanged. When the tooth side clearance is reduced to 40 μ m, severe collisions occur between the meshing teeth, causing a sharp increase in the peak meshing force. In the frequency-domain signal, the frequency components are dominated by the meshing frequency itself, and the sideband frequencies almost disappear.



a) Time-domain signal of the meshing force F_{spi1} b) Frequency-domain signals of meshing force F_{spi1} **Fig. 8.** Effect of changing the tooth side clearance on meshing force

In practical applications, gear manufacturing precision and assembly errors inevitably introduce transmission errors. Additionally, factors such as processing variations and tooth wear contribute to increased gear transmission errors. To investigate the influence of transmission errors on gear meshing forces, this paper utilize unbalance loading mode 2 as an example. The load is set to 10% of the rated capacity, and the amplitudes of gear pair transmission errors are varied to $2~\mu m$, $20~\mu m$, and $50~\mu m$. The results are shown in Fig. 9.

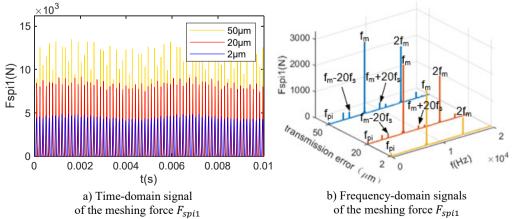


Fig. 9. Effect of transmission error on gear meshing force

Fig. 9 demonstrates that transmission error significantly influences gear meshing characteristics. Smaller fluctuations in transmission error result in reduced gear tooth excitation, thereby minimizing variations in gear meshing force. Conversely, an increasing in the amplitude of transmission error fluctuations correlates with heightened fluctuations in gear meshing force. When the transmission error excitation on gear teeth reaches a critical level during meshing, it can cause teeth impacts. As transmission error further increases, the amplitude of gear meshing force amplitude gradually rises. In cases without tooth collisions, the frequency-domain signal of meshing force primarily consists of the meshing frequency and its harmonics. Conversely, tooth collisions introduce modulated side-band frequencies, with their amplitudes increasing in tandem with the transmission error fluctuations.

5. Analysis of kinetic and strain energy and load equalization factor for planetary gear system

The strain energy and kinetic energy of a planetary gear system are important forms of energy in describing its operation and design, while the strain energy affects the stability and life of the system. The strain energy of a planetary gear system is primarily caused by the elastic deformation of the rotor, with the most significant contribution comes from the bending strain energy, expressed as:

$$E_s = \frac{1}{2} x^T k x, \tag{26}$$

where E_s is the bending adaptive energy, k, x are the bending stiffness matrix and bending displacement matrix of the rotor, respectively.

The kinetic energy of the planetary gear system affects the dynamic response and energy conversion efficiency of the system, the kinetic energy of the star gear system mainly comes from its rotating parts, which is expressed as:

$$E_d = \frac{1}{2} v^T m v, \tag{27}$$

where E_d is the kinetic energy, m, v are the mass matrix and velocity matrix of the rotor system.

To investigate the impact of load and system parameters on the kinetic and strain energy of the planetary gear system, two operating conditions were examined: ideal and actual states. In the ideal state, the system operates at the specified speed and load without transmission errors and with minimal tooth side clearance. In the actual state, the system operates under the rated load with transmission errors and a large tooth side clearance. according to Eq. (26), the kinetic energy of each gear shaft over one cycle was calculated for both the ideal and actual states. To facilitate identification, the gear shafts are defined by the names of their respective gears. The results are shown in Fig. 10, from which it can be observed that in the ideal state, the kinetic energy of the individual gear shafts increases, particularly for the ring gear and the sun gear, where the kinetic energy is relatively non-uniform.

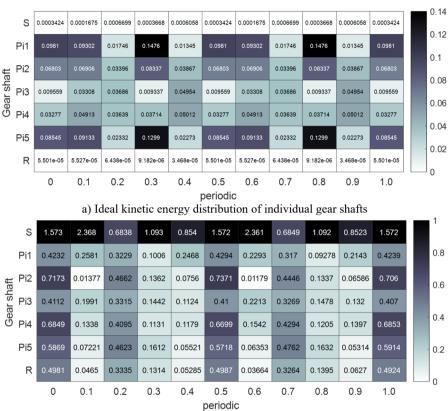
To thoroughly investigate the influence of each system parameter on the kinetic energy, the kinetic energy of the sun gear is taken as an example, and the effects of different factors in the actual state is analyzed, and the results are shown in Fig. 11.

Fig. 11(a) shows the variation of kinetic energy of the shaft of the sun gear over one cycle by changing the tooth side clearance to 20 μ m, 30 μ m and 40 μ m, respectively. From the figure, it can be observed that the kinetic energy of the sun gear decreases as decreases as the tooth side clearance increases, and the fluctuation of the kinetic energy decreases with the increase of the tooth side clearance.

Fig. 11(b) shows the variation of kinetic energy of the sun gear shaft over one cycle by

changing the transmission error amplitude to 2 μ m, 20 μ m and 50 μ m respectively. From the figure, it can be observed that the kinetic energy fluctuation will increase with the transmission error amplitude as the transmission error amplitude increases, but the degree of influence is smaller compared with that in Fig. 11(a).

Fig. 11(c) shows the variation of the kinetic energy of the sun gear shaft over one cycle by changing the load torque amplitude to 50 % of the rated load, 100 % of the rated load, and 200 % of the rated load, respectively. It can be observed that both the amplitude and fluctuation of the kinetic energy of the sun gear increase with the load.



b) Kinetic energy distribution of individual gear shafts in the actual state Fig. 10. Kinetic energy distribution of each gear shaft in two states

Strain energy is an important factor in the process of optimizing the design of gears and related components. Analyzing the distribution of strain energy in a planetary gear system and the effect of nonlinear factors on it is essential for understanding of the system performance. In this paper, the strain energy analysis of the planetary gear system supports is carried out. According to Eq. (27), the strain energy of each support over one cycle under two working conditions is calculated, and the results are shown in Fig. 12. Since most of the strain energy of the ring gear is stored within itself, the strain energy of the support of the ring gear is smaller compared to that of the planetary gear shaft supports. Additionally, the stiffness of the right bearing of the planetary gear shaft is smaller than that of the left bearing, resulting in higher potential energy in the right bearing compared to the left. Comparing Fig. 12(a) and Fig. 12(b), it can be observed that the fluctuation of the strain energy in the actual state is larger than that in the ideal state. To better understand the influence of the system parameters on the strain energy, the strain energy of the right bearing (Pi1y) of the sun gear shaft is analyzed as an example, and the results are shown in Fig. 13.

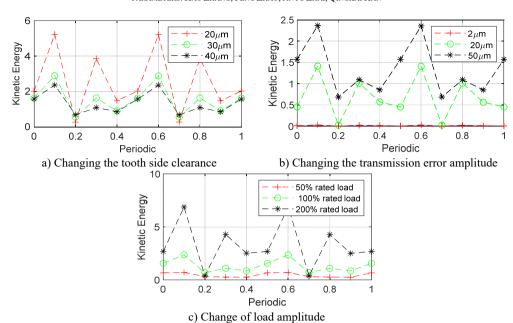


Fig. 11. Effect of each system parameter on the kinetic energy of the sun gear axis

Pi1z	4.298	4.299	4.288	4.28	4.287	4.298	4.299	4.288	4.28	4.287	4.298
Pi1v	6.067	6.061	6.047	6.044	6.057	6.067	6.061	6.047	6.044	6.057	6.067
Pi2z	4.655	4.66	4.741	4.786	4.732	4.655	4.66	4.741	4.786	4.732	4.655
Pi2y	6.527	6.539	6.616	6.651	6.595	6.527	6.539	6.616	6.651	6.595	6.527
- 1	4.373	4.356	4.26	4.217	4.285	4.373	4.356	4.26	4.217	4.285	4.373
Pi3z Pi3y	6.12	6.1	6.01	5.973	6.04	6.12	6.1	6.01	5.973	6.04	6.12
Pi4z	4.493	4.507	4.606	4.654	4.583	4.493	4.507	4.606	4.654	4.583	4.493
Pi4z Pi4z Pi4y	6.332	6.345	6.434	6.476	6.412	6.332	6.345	6.434	6.476	6.412	6.332
Pi5z	4.622	4.609	4.555	4.534	4.576	4.622	4.609	4.555	4.534	4.576	4.622
Pi5y	6.427	6.419	6.374	6.352	6.386	6.427	6.419	6.374	6.352	6.386	6.427
Rz	4.735e-08	6.57e-08	8.809e-08	8.26e-08	5.5e-08	4.735e-08	6.57e-08	8.809e-08	8.26e-08	5.5e-08	4.735e-08
Ry	6.821e-08	8.526e-08	1.127e-07	1.119e-07	8.172e-08	6.821e-08	8.526e-08	1.127e-07	1.119e-07	8.172e-08	6.821e-08
	0	0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9	1.0
	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	D: 4 '1	·. c			Periodic	,.	1	1 '1	1 1	.,.
	a).	Distribu	tion of s	strain en	ergy at	each su	pporting	g noae u	naer 1ae	eai cond	itions
Pi1z	4.276	4.274	4.406	4.596	4.516	4.185	3.947	4.045	4.432	4.658	4.53
Pi1y	6.063	6.088	6.223	6.404	6.315	5.953	5.673	5.756	6.179	6.486	6.419
Pi2z	5.087	4.857	4.806	4.913	4.912	4.705	4.653	4.886	5.222	5.235	4.825
Pi2y	6.92	6.867	6.87	6.899	6.782	6.512	6.464	6.745	7.169	7.298	6.966
용 Pi3z	4.187	3.809	3.63	3.776	3.999	4.077	4.018	4.012	4.089	4.032	3.808
Pi3z Pi3y	5.986	5.637	5.445	5.574	5.79	5.867	5.821	5.832	5.937	5.901	5.659
t. Q Pi4z	4.778	4.466	4.559	4.986	5.25	5.068	4.812	4.758	4.917	4.959	4.76
Support Bi4s Bi4s	6.746	6.369	6.4	6.813	7.14	7.077	6.893	6.784	6.8	6.711	6.492
Pi5z	4.121	4.05	4.239	4.598	4.711	4.416	4.026	3.935	4.208	4.507	4.541
Pi5y	5.981	5.833	5.971	6.31	6.484	6.298	5.944	5.788	5.923	6.115	6.168
Rz	0.007749	0.0007506	0.003006	0.008072	0.006921	0.003074	0.002085	0.001192	0.0001305	0.005293	0.01411
Ry	0.00869	0.0008112	0.003418	0.00917	0.007921	0.003679	0.002586	0.001482	0.0001451	0.005803	0.01565
'	0	0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9	1.0
						Periodic					

b) Distribution of strain energy at each supporting node in the actual state Fig. 12. Distribution of strain energy for each support in both states

Fig. 13(a) shows the variation of strain energy over one cycle by hanging the tooth side clearance to 20 μ m, 30 μ m and 40 μ m. From the figure, it can be observed that the fluctuation of strain energy decreases as the tooth side clearance increases, while the amplitude of strain energy remains relatively unchanged.

Figure 13(b) shows the variation of strain energy over one cycle by changing the transmission error amplitude to 2 μ m, 20 μ m and 50 μ m. From the figure, it can be observed that as the transmission error amplitude increases, the fluctuation of potential energy increases, and the amplitude of potential energy also rises to some extent. Compared with Fig. 12(a), the degree of influence is more significant.

Fig. 13(c) shows the variation of strain energy over one cycle by varying the load torque amplitude to 50 % of rated load, 100 % of rated load, and 200 % of rated load torque, respectively, and the amplitude of adaptive energy increases dramatically as the load increases and fluctuation increases.

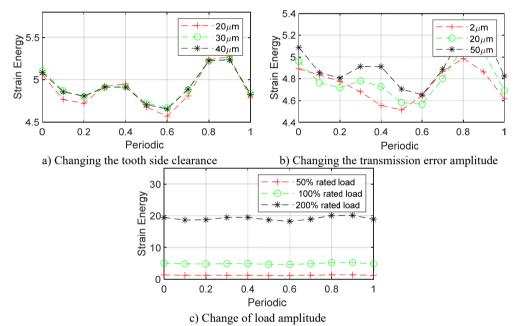


Fig. 13. Effect of system parameters on strain energy of planetary gear right support

In a planetary gear system, the load applied between each pair of meshing gear teeth exhibits non-uniform distribution. To quantify this, we introduce the load distribution coefficient, also known as the load sharing factor. The load sharing coefficient for the planetary gear system is defined as the moment of the meshing force divided by the average meshing force ratio. Ideally, uniform load distribution across the entire system results in a load sharing factor of 1, indicating optimal load equalization. Deviations from this ideal lead to varying degrees of load imbalance. The left and right external meshing and internal meshing equalization coefficients for each gear pair are defined as b_{spin} , b'_{spin} , b_{rpin} , b'_{rpin} , where N = 1, 2, 3, 4, 5, and are expressed as:

$$\begin{cases}
B_s = \max(b_{spiN}), & B'_s = \max(b'_{spiN}), \\
B_r = \max(b_{rpiN}), & B'_r = \max(b'_{rpiN}).
\end{cases}$$
(28)

Fig. 14 illustrates the variation in load sharing coefficients for the left and right sides, as well as internal and external meshes of the planetary gear system, under two operating conditions: operating speed and rated load. As shown in Fig. 14(a), under ideal condition, the load sharing

coefficients for the left and right sides, as well as internal and external meshing, exhibit cyclic variations. Although peak fluctuations remain relatively stable, differences in amplitudes are observed due to varying meshing phases. Conversely, in the actual operating state (see Fig. 14(b)), all load sharing coefficients increase significantly, accompanied by larger peak fluctuations. Overall, the load sharing coefficients for internal and external meshing on the left side are higher than those on the right side. This discrepancy is primarily attributed to the greater support stiffness of the left gear, which results in less effective load balancing.

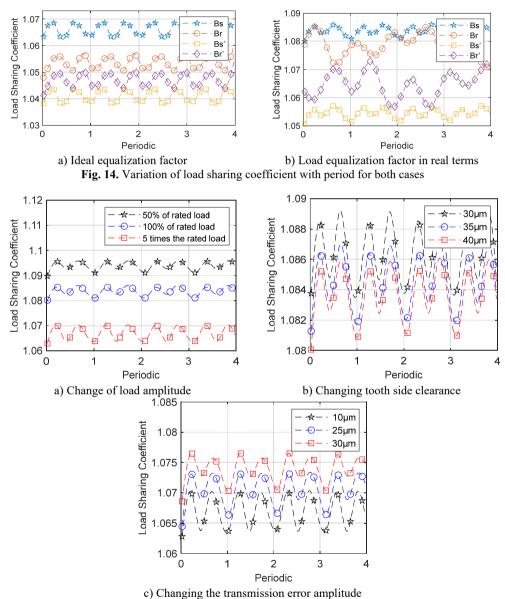


Fig. 15. Effect of individual system parameters on load sharing coefficient

To thoroughly examine the impact of each system parameter on load balancing properties, adjustments were made to tooth side clearance, load, and transmission error using the method of controlling variables. As an illustration, analyzed the external mesh load sharing coefficient B_s

was analyzed, and the outcomes are presented in Fig. 15.

In Fig. 15(a), the graph depicts the variation of the external mesh equalization coefficient over time for different loads. As the load increases, load sharing coefficient gradually converges to 1, indicating that higher loads improve load balancing properties.

In Fig. 15(b), the graph shows how the external mesh load sharing coefficient changes with cycle time for varying degrees of tooth side clearance. As the tooth side clearance increases, the equalization coefficient decreases gradually. However, the impact of the tooth side clearance is less significant compared to the influence of load.

Fig. 15(c) illustrates how transmission error affects load sharing coefficient. As the amplitude of transmission error increases, the fluctuation in the load sharing coefficient decreases, but overall equalization properties worsen, suggesting that smaller transmission errors improve system equalization properties.

Fig. 16 presents load sharing coefficient under the influence of each system parameter. It is evident that load has the greatest impact on load sharing coefficient: higher loads correlate with smaller coefficients, indicative of better equalization effects. Tooth side clearance minimally impacts on load sharing coefficient, whereas transmission error significantly impacts: an increase in transmission error amplifies the load sharing coefficient, worsening the load equalization effects.

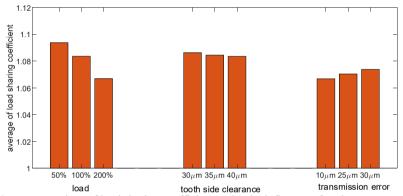


Fig. 16. Average values of load sharing coefficient under the influence of each system parameter

6. Conclusions

This study develops a nonlinear time-varying model using the finite element nodal method, and provides a comprehensive analysis of the impact of load and system parameters on meshing force, system kinetic energy, strain energy, and the load sharing coefficient of the planetary gear system, The study draws the following key conclusions:

- 1) Unbalanced vibration in the planetary gear system exacerbates the collision between gear teeth and increases the amplitude of the gear meshing force. By analyzing the side-band frequency intervals, the mechanism of the unbalance-induced meshing side-band modulation is revealed. The appearance and amplitude of the sideband frequency are related to the gear collision. As the gear collision intensifies, the amplitude of the sideband frequency increases.
- 2) When the sum of the internal and external excitations on a pair of meshing gears exceeds the tooth pressure required to maintain meshing, the gears may disengage, resulting in impacts between the teeth during the meshing process. Increasing the load or reducing the transmission error can mitigate these impacts. In the case of unilateral impacts during gear meshing, when the tooth side clearance is reduced below a certain threshold, the unilateral collision between the gear teeth will be transformed into a bilateral collision, leading to a significant increase in the peak value of the sideband frequency.
 - 3) The distribution of kinetic energy is significantly influenced by tooth side clearance.

transmission error, and load torque. Increasing the tooth side clearance can reduce kinetic energy fluctuations and improve system stability. In contrast, transmission error has a relatively minor impact on kinetic energy. Although changes in tooth side clearance a limited effect on strain energy, increased transmission error and load torque significantly elevate both the amplitude and fluctuation of strain energy, thereby affecting system longevity and stability.

4) The influence of tooth side clearance on the load sharing coefficient is relatively minor. Although an increase in tooth side clearance may slightly reduce load sharing coefficient, this effect is not substantial. In conversely, transmission error has a pronounced impact on load equalization, with higher transmission errors exacerbating fluctuations in the load sharing coefficient. Load torque has the most significant effect on load sharing coefficient; increasing load torque can significantly improve the load sharing coefficient of the system.

Acknowledgements

This research project is supported by the Aeronautical Science Foundation of China (No. 2022Z042063001), the 2024 Fundamental Research Project of the Educational Department of Liaoning Province (No. LJ212410154038) and Natural Science Foundation of Liaoning Province (2023-MS-301).

Data availability

The datasets generated during and/or analyzed during the current study are available from the corresponding author on reasonable request.

Author contributions

Hao Zhang: conceptualization. Peng Zhao: methodology. Jingyu Zhai: supervision. Qingkai Han: project administration.

Conflict of interest

The authors declare that they have no conflict of interest.

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